



# TECHNICAL PUBLICATIONS INDEX

SORTED BY PART NUMBER

UPDATED APRIL 4, 2024

## CONTACT INFORMATION

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### ABBREVIATIONS USED IN THIS INDEX:

ACMM = ABBREVIATED COMPONENT MAINTENANCE MANUAL  
ATLAS = ABBREVIATED TEST LANGUAGE FOR ALL SYSTEMS  
CMM = COMPONENT MAINTENANCE MANUAL  
GSE = GROUND SUPPORT EQUIPMENT MANUAL  
N/A = NOT APPLICABLE  
OHM = OVERHAUL MANUAL  
QRG = QUICK REFERENCE GUIDE  
SB = SERVICE BULLETIN  
SL = SERVICE INFORMATION LETTER  
TR = TEMPORARY REVISION  
TS = TEST SPECIFICATION  
TSDP = TECHNICAL SUPPORT AND DATA PACKAGE  
TSG = TROUBLESHOOTING GUIDE

NOTE:  
AIRCRAFT MODELS LISTED IN THIS INDEX ARE FOR REFERENCE ONLY. REQUIREMENTS FOR TECHNICAL DOCUMENTATION SHOULD BE DETERMINED FROM AIRCRAFT OPERATOR RECORDS, NOT FROM THIS INDEX.

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HYDRO-AIRE

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
00-205E	HYTROL ANTI-SKID BRAKING SYSTEM	COVAIR 990	CMM	32-40-0	0	11/15/1961	00-205E
00-277	HYTROL MARK II ANTI-SKID BRAKING SYSTEM	??	MM	32-40-0	0	7/15/1967	00-277
00-793-1	HYDRAULIC SYSTEM	AERMACCHI MB339	OP HBK	(No ATA Number)	0	5/4/1984	OP-HBK00-793-1
00-841-1	HYDRAULIC SYSTEM	EMB-120	OP HBK	(No ATA Number)	2	11/1/1989	OP-HBK00-841-1
			PS DATA	(No ATA Number)	0	2/24/1986	PS-DATA00-841-1
00-845-1	SYSTEM INSTALLATION MARK III GA POWER BRAKE AND SKID CONTROL SYSTEM	BEECH 1900	OP HBK	(No ATA Number)	1	6/15/1992	OP-HBK00-845-1
00-885-1	TBD	DHC-8	PS DATA	(No ATA Number)	0	12/1/1984	PS-DATA00-885-1
00-885-1A	TBD	DHC-8	OP HBK	(No ATA Number)	0	2/1/1985	OP-HBK00-885-1A
00-893	TBD	B-737-300	PS DATA	(No ATA Number)	0	3/1/1986	PS-DATA00-893
00-901	TBD	ATR-42, ATR-72	OP HBK	(No ATA Number)	0	11/1/1985	OP-HBK00-901
			PS DATA	(No ATA Number)	0	3/2/1986	PS-DATA00-901
00-951	TBD	AMX	OP HBK	(No ATA Number)	0	3/18/1985	OP-HBK00-951

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.							
2-8332-1, 2-8332-2	BRAKE SYSTEM CONTROL UNIT (BSCU)	787-8	CMM	32-42-06	7	6/21/2023	CMM2-8332	R						
			SB	2-8332-1-32-2	0	10/31/2018	2-8332-1-32-2							
			<i>This Service Bulletin provides mitigation for occurrences of BRAKE INDICATION that may get reported to the flight deck during takeoff shortly after ground-to-air transition.</i>											
			SB	2-8332-2-32-1	1	3/14/2017	2-8332-2-32-1							
			<i>To provide instructions to inspect PPCM PCB assemblies A2, P/N 8-969101-03 with date code 1418</i>											
			SB	SB2-8332-1-32-1	1	3/7/2019	SB2-8332-1-32-1							
			<i>To provide instructions to convert P/N 2-8332-1 BSCUs to P/N 2-8332-2</i>											
			TRD	N/A	C	5/31/2012	TRD-142-143-02							
2-8350-2	MAIN LANDING GEAR (MLG) AXLE REMOTE DATA CONCENTRATOR (ARDC) MOUNTING ADAPTER ASSEMBLY	787	CMM	32-42-08	0	8/31/2010	CMM2-8350-2							
			2-8351-2, -3	NOSE LANDING GEAR (NLG) AXLE REMOTE DATA CONCENTRATOR (ARDC) MOUNTING ADAPTER ASSEMBLY	787-8, -9	CMM	32-49-01	2	6/5/2015	CMM2-8351-2				
						6-0488-2	MAGNET RING ASSEMBLY	787-8	CMM	32-42-07	2	3/14/2014	CMM6-0488	
						09-005, 09-005A, 09-005B, 09-005C	ROTOR BRAKE	SIKORSKY S76	CMM	66-50-01	7	10/25/2011	CMM09-005	
									SB	09-005A-32-98	0	2/24/1986	09-005A-32-98	
<i>Replace low pressure switch to provide a greater difference between pressure indication and initial rotor brake contact pressure of system.</i>														
SB	09-005B-32-133	0	2/24/1986	09-005B-32-133										
<i>Increase system volume from 10 cubic inch to 16 cubic inch maximum.</i>														

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A-60203-5001, -5501, 5503, 5505, 5507, 5509	ASPIRATOR AND PRESSURE-SENSITIVE SHUTOFF VALVE	DC-8	OHM	29-20-13	0	5/15/1979	OHM_A60203-5001
A-62080, -1, -2, -3, -4, -5, -6	HYDRAULIC PRIORITY VALVE	B-707/727/757	CMM	29-20-02	5	7/31/2007	CMM_A-62080
A-63002, A-63003	BALANCED RELIEF VALVE ASSEMBLY	DC-8	CMM	29-20-12	0	2/6/2009	CMM_A-63002
A-63004, A-63005	BALANCED RELIEF VALVE ASSEMBLY	DC-8	CMM	29-20-01	0	1/27/2009	CMM_A-63004
D-26-76 LH; D-26-75 RH	SHUT OFF VALVE	???	SB <i>(Description not available)</i>	(No ATA number)	0	9/1/1976	D-26-75 RH -- D-26-76
100-033	TBD	BAE JETSTREAM 41	PS DATA	(No ATA Number)	0	5/15/1991	PS-DATA100-033
11-013	TBD	CESSNA CITATION 1	OP HBK	(No ATA Number)	0	1/1/1975	OP-HBK11-013
11-017	TBD	METROLINER	OP HBK	(No ATA Number)	0	1/8/1981	OP-HBK11-017
11-079	TBD	BJET400	OP HBK	(No ATA Number)	2	5/15/1992	OP-HBK11-079
11-091-1	TBD	SAAB 340	OP HBK PS DATA	(No ATA Number) (No ATA Number)	1 0	3/16/1984 12/1/1984	OP-HBK11-091-1 PS-DATA11-091-1
13-00122-100	TBD	King Air B200	SB <i>(Description not available)</i>	(No ATA Number)	0	1/1/1901	N/A
			SB	13-00122-100-1	0	11/14/2007	13-00122-100-1 <i>Installs Master Cylinder retract control orifice fitting assemblies in the left and right MC lines that connect the co-pilot MC to the Power Brake and Skid Control System.</i>
138-007	EMERGENCY PARK BRAKE VALVE (EPBV)	ERJ 170/190	CMM	32-44-02	0	2/17/2020	CMM138-007

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138-025	SHUT OFF VALVE (SOV)	C919	CMM	32-41-04	1	6/6/2022	CMM138-025
138-029500	CHECK VALVE	C919	ACMM	32-43-02	0	8/30/2019	ACMM138-029500
138-031500	ONE-WAY RESTRICTOR VALVE	C919	ACMM	32-41-08	0	9/13/2019	ACMM138-031500
138-033500, 138-033500-01	PARKING BRAKE VALVE (PBV)	C919	ACMM	32-43-04	2	3/1/2023	ACMM138-033500
138-047500	THERMAL RELIEF VALVE	C919	ACMM	32-43-05	0	9/13/2019	ACMM138-047500
138-057	VALVE COMPENSATOR ASSEMBLY	PC-21	CMM	32-41-87	0	7/22/2022	CMM138-057
138-065	EMERGENCY PARK BRAKE VALVE	E2	CMM	32-44-58	0	2/15/2019	CMM138-065
138-089500	THERMAL RELIEF VALVE	E2	CMM	32-44-60	0	7/13/2018	CMM138-089500
138-121	SHUT OFF VALVE (SOV)	CESSNA 700	CMM	32-44-70	0	5/12/2021	CMM138-121
138-125	SHUTTLE VALVE	???	CMM	32-44-72	0	3/2/2021	138-125
138-153, 138-153-1	MANIFOLD HYDRAULIC FUSE	VARIOUS	CMM	29-20-78	0	3/15/2023	CMM138-153
138-161	SHUTOFF VALVE (SOV)	???	CMM	32-44-53	0	12/12/2019	CMM138-161
138-221	SHUTOFF VALVE ASSEMBLY	BD-700	CMM	32-44-74	0	12/12/2023	CMM138-221 R
140-007	TRANSDUCER ASSEMBLY	CESSNA 650	CMM	32-42-68	3	5/15/2016	CMM140-007
140-025-1, 140-025-2	WHEEL SPEED TRANSDUCER (WST)	B-737-700/-800,757,767,747-400/777	CMM	32-42-54	13	8/15/2017	CMM140-025
			SB	140-025-1-32-141	0	12/2/1985	140-025-1-32-141
<i>Increased strength locking plate to minimize effects of carbon brake vibration levels.</i>							

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140-031	WHEEL SPEED TRANSDUCER	BJET400	CMM	32-42-69	1	6/27/1999	CMM140-031
			SB	140-031-32-1	0	11/15/1993	140-031-32-1
				<i>To assure proper engagement between hubcap drive clip and transducer drive coupling.</i>			
140-041, 140-04120, 140-04120-1	WHEEL SPEED TRANSDUCER (WST)	SAAB 340	CMM	32-43-15	3	3/15/2016	CMM140-041
			SB	140-041-32-1	0	1/3/1996	140-041-32-1
				<i>Modify hubcap to incorporate new drive clip.</i>			
			SB	140-041-32-2	0	5/25/1998	140-041-32-2
				<i>Replacement procedure if spacer is installed.</i>			
140-043, 140-04320	TRANSDUCER ASSEMBLY, HUBCAP ASSEMBLY	EMB-120	CMM	32-42-82	2	3/31/2016	CMM140-043
140-047, 140-04715, 140-04730	WHEEL SPEED TRANSDUCER (WST) AND HUBCAP ASSEMBLY	BEECH 1900	CMM	32-42-98	1	6/30/2007	CMM140-047
140-057, 140-057-1, 140-05722, 140-05724	WHEEL SPEED TRANSDUCER (WST)	DHC-8	CMM	32-42-77	5	5/23/2016	CMM140-057
			SB	140-057-32-148	0	9/11/1986	140-057-32-148
				<i>To alleviate problems of damaging drive clip inside hubcap P/N 140-05720 encountered when mating hubcap to wheel speed transducer drive coupling.</i>			
140-071, 140-07120	WHEEL SPEED TRANSDUCER and HUBCAP ASSEMBLY	Alenia C27J (ATR 42/72)	CMM	32-42-87	3	12/28/2006	CMM140-071
140-139	WHEEL SPEED TRANSDUCER	CBA 123	CMM	32-43-17	0	9/1/1990	CMM140-139
140-149, 140-04114, 140-14920	WHEEL SPEED TRANSDUCER (WST) ASSEMBLY	BAE JETSTREAM 41	CMM	32-43-33	2	3/15/2016	CMM140-149

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140-153, 140-15320	WHEEL SPEED TRANSDUCER (WST) ASSEMBLY, HUBCAP ASSEMBLY	CRJ-100/200	CMM	32-43-26	6	6/1/2023	CMM140-153	R
140-159	WHEEL SPEED TRANSDUCER	SAAB 2000	CMM	32-43-35	0	10/15/1991	CMM140-159	
			SB	140-159-32-1	0	1/3/1996	140-159-32-1	
			<i>Modify hubcap to incorporate new drive clip</i>					
140-177	WHEEL SPEED TRANSDUCER (WST)	B-777	CMM	32-43-59	3	3/15/2016	CMM140-177	
140-179, 140-17915, 140-17920	WHEEL SPEED TRANSDUCER (WST)	CESSNA 750	CMM	32-43-56	3	1/26/2024	CMM140-179	R
			SB	140-179-32-1	0	1/11/1999	140-179-32-1	
			<i>Apply Loctite adhesive and sealant to the threads of the transducer housing (if applicable) and adding lockwire to the assembly for positive retention</i>					
140-181, 140-18120	WHEEL SPEED TRANSDUCER (WST) AND HUBCAP ASSEMBLY	CL604	CMM	32-43-51	2	3/15/2016	CMM140-181	
140-183, 140-18320	WHEEL SPEED TRANSDUCER (WST)	LEAR 45, AF-8	CMM	32-43-45	1	1/31/2010	CMM140-183	
140-197	WHEEL SPEED TRANSDUCER (WST)	EMB-145, 145-AEW&C	CMM	32-41-03	2	7/15/2006	CMM140-197	
			SB	140-197-32-1	0	2/24/1999	140-197-32-1	
			<i>(Description not available)</i>					
140-199, 140-199-1	WHEEL SPEED TRANSDUCER (WST)	GULFSTREAM V	CMM	32-45-03	2	3/15/2016	CMM140-199	
			SB	140-199-1-32-01	0	11/16/2006	140-199-1-32-01	
			<i>Modification to convert WST from 140-199 to the 140-199-1 configuration. The new design reduces unwanted deflection and eliminates erroneous wheel speed signals from being generated at the antiskid dropout speed.</i>					

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140-209-1,-2, -3, -4, 140-20920-1	WHEEL SPEED TRANSDUCER (WST)	GLOBAL EXPRESS	CMM	32-43-66	10	1/15/2016	CMM140-209-1
			SB	140-209-2-32-1	1	5/18/2002	140-209-2-32-1
			<i>Modification of Wheelspeed Transducer P/N 140-209,-1,-2 (Rev C and Earlier) to P/N 140-209-2 Rev D configuration.</i>				
			SB	140-209-3-32-1	0	2/15/2009	140-209-3-32-1
<i>Modification of Wheelspeed Transducer, P/N 140-209-2 (Rev D) to P/N 140-209-3.</i>							
140-221	WHEEL SPEED TRANSDUCER (WST) AND ADAPTER ASSEMBLY	PREMIER 1	CMM	32-43-73	0	10/27/2005	CMM140-221
140-227, 140-22710, 140-22711	WHEEL SPEED TRANSDUCER (WST) AND ADAPTER ASSEMBLY	B-717-200	CMM	32-45-05	6	10/24/2022	CMM140-227
140-243	WHEELSPEED TRANSDUCER (WST)	DORNIER 728JET	CMM	32-43-16	0	3/15/2002	CMM140-243
140-247, -1, -2; 140-24720	WHEEL SPEED TRANSDUCER (WST) ADAPTER ASSEMBLY AND HUBCAP ASSEMBLY	BD-100	CMM	32-43-49	2	6/30/2009	CMM140-247
140-261, 140-26120	WHEEL SPEED TRANSDUCER (WST)	PILATUS PC21	CMM	32-41-10	0	7/1/2015	CMM140-261
140-265	WHEEL SPEED TRANSDUCER (WST)	ERJ-170-/190	CMM	32-41-11	3	1/31/2018	CMM140-265
140-271	WHEEL SPEED TRANSDUCER (WST) ASSEMBLY	ERJ-170	CMM	32-41-33	1	6/13/2018	CMM140-271
140-27120	HUBCAP ASSEMBLY	ERJ-170-/190	CMM	32-41-13	3	4/15/2016	CMM140-27120
140-287	WHEEL SPEED TRANSDUCER (WST)	MD-10	CMM	32-45-14	2	2/26/2020	CMM140-287

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140-289	WHEEL SPEED & SPOILER TRANSDUCER ASSEMBLY, DUAL WHEEL SPEED TRANSDUCER (WST) ASSEMBLY, ADAPTER ASSEMBLY	MD-10	CMM	32-45-13	1	8/25/2017	CMM140-289
140-29120	HUBCAP ASSEMBLY	ERJ-190	CMM	32-41-14	2	2/20/2013	CMM140-29120
140-29720	TBD	CX/PX	CMM	32-45-18	0	11/15/2004	CMM140-29720
140-301	TBD	EMB 314 (SUPER TACANO)	CMM	32-43-86	0	5/14/2010	CMM140-301
140-329, 140-32911	WHEEL SPEED TRANSDUCER (WST)	HA-420	CMM	32-45-95	0	3/11/2021	CMM140-329
140-343-03	MAIN LANDING GEAR AXLE ADAPTER ASSEMBLY	COMAC C919	CMM	32-41-09	0	11/23/2021	CMM140-343-03
140-363-01	WHEEL SPEED TRANSDUCER (WST)	Pilatus PC-21	CMM	32-41-82	0	5/15/2019	CMM140-363-01
140-369	WHEEL SPEED TRANSDUCER (WST)	E2	CMM	32-41-48	0	7/13/2018	CMM140-369
140-36920	HUBCAP ASSEMBLY	E2	CMM	32-41-50	0	7/13/2018	CMM140-36920
140-383, 140-17915	WHEEL SPEED TRANSDUCER (WST)/ADAPTER ASSEMBLY	CESSNA 700	CMM	32-45-90	0	4/4/2021	140-383
140-40920	HUBCAP ASSEMBLY	UTAS 1032	ACMM	32-45-81	0	9/24/2020	ACMM140-40920
142-015, -1	DIGITAL MARK IV CONTROL BOX	CESSNA 550 BRAVO	CMM	32-45-60	1	2/15/2008	CMM142-015
			SB	142-015-32-1	0	3/26/2007	142-015-32-1
			<i>To prevent unexpected low speed brake release command.</i>				

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142-021	DIGITAL MARK IV CONTROL BOX	CESSNA CITATION 560 ENCORE	CMM	32-41-21	1	3/25/2010	CMM142-021
142-027-1	ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	B-767-400	CMM	32-43-78	1	8/27/2004	CMM142-027
			ATLAS	N/A	R	1/8/2001	ATL757
			TR	TR32-43-78-01	0	11/21/2016	TR32-43-78-01
			<i>Update IPL Figure 2, Item 75 and add Item -75A.</i>				
			TR	TR32-43-78-02	0	11/21/2016	TR32-43-78-02
<i>Update IPL Figure 3, Item 195 and add Item -195A.</i>							
			TR	TR32-43-78-03	0	11/21/2016	TR32-43-78-03
<i>Update IPL Figure 4, Item 50 and add Item -50A.</i>							
142-031, -1, -3, -4	DIGITAL MARK IV CONTROL BOX ASSEMBLY	CESSNA	CMM	32-41-09	2	1/25/2012	CMM142-031
			SB	142-031-32-1	0	4/4/2007	142-031-32-1
<i>To prevent unexpected low speed brake release command.</i>							
142-045, 142-045-1, 142-045-2	HYTROL MARK V BRAKE CONTROL UNIT (BCU)	BD-100	CMM	32-43-05	7	1/24/2024	CMM142-045 R
			SB	142-045-1-32-1	0	9/23/2008	142-045-1-32-1
			<i>Instruction to return the BCU's to Hydro Aire for repair and verification testing.</i>				
			SB	142-045-1-32-2	0	4/19/2012	142-045-1-32-2
<i>To provide instruction to check if alternate ASICs have already been installed (as needed).</i>							
			SB	142-045-1-32-3	0	11/8/2013	142-045-1-32-3
<i>This product improvement addresses aircraft performance during hydraulic system pressure loss, with a related improvement in park brake pressure maintenance.</i>							
142-059, 142-059-01	DIGITAL MARK IV SCBS CONTROL BOX	PILATUS PC-21	CMM	32-41-12	1	7/9/2021	142-059

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142-07114; 142-071141; 142-071142	BRAKE CONTROL MODULE (BCM)	ERJ-170	CMM	32-41-07	6	5/10/2023	CMM142-07114	R	
			SB	170-32-142-07114-01	0	1/28/2005	170-32-142-07114-01		
			<i>Modification to convert P/N 142-07114 BCM Assembly to P/N 142-071141 BCM Assembly</i>						
			SB	170-32-142-07114-02	1	7/28/2009	170-32-142-07114-02		
			<i>Modification to convert P/N 142-07114 or 142-071141 Brake Control Module (BCM) Assembly to P/N 142-071142 BCM Assembly</i>						
			SB	170-32-142-071142-01	0	11/11/2009	170-32-142-071142-01		
			<i>Modification to convert P/N 142-071141 and 142-071142 Brake Control Module (BCM) Assembly to 142-091142 BCM Assembly</i>						
			SB	190-32-142-09114-02	1	8/1/2022	190-32-142-09114-02		
<i>To eliminate noise on the ALE line that causes intermittent microcontroller resets.</i>									
			SB	190-32-142-09114-03	0	2/8/2023	190-32-142-09114-03		
<i>(Description not available)</i>									
			SB	190-32-142-091142-01	0	10/1/2014	190-32-142-091142-01		
<i>BCM software modification to P/N 142-091143.</i>									
			TR	TR32-41-07-01	0	5/8/2018	TR32-41-07-01		
<i>Revises CMM 32-41-07, Revision 5, Highlights Page 1.</i>									
142-07117; 142-071171	AUTOBRAKE CONTROL MODULE (ACM)	ERJ-170/-190	CMM	32-41-02	4	5/10/2023	CMM142-07117	R	
			SB	170-32-142-07117-01	0	1/3/2008	170-32-142-07117-01		
			<i>To improve autobrake disengagement to avoid abrupt changes in aircraft deceleration. To improve autobrake disarm logic to diminish occurrences of spurious aural warning.</i>						
			SB	170-32-142-07117-02	0	1/11/2019	170-32-142-07117-02		
<i>This modification will eliminate noise on the ALE line that may cause inadvertent ABS disarms.</i>									
			SB	170-32-142-07117-03	0	2/8/2023	170-32-142-07117-03		
<i>(Description not available)</i>									
142-073	MARK IV DIGITAL ANTISKID CONTROL UNIT (DACU)	CESSNA CITATION 680 SOVEREIGN	CMM	32-41-26	3	7/21/2022	CMM142-073		

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142-09114, 142-091141, 140-091142, 142-091143	BRAKE CONTROL MODULE	ERJ-190	CMM	32-41-21	6	4/25/2023	CMM142-09114	R
			SB	170-32-142-071142-01	0	11/11/2009	170-32-142-071142-01	
			<i>Modification to convert P/N 142-071141 and 142-071142 Brake Control Module (BCM) Assembly to 142-091142 BCM Assembly</i>					
			SB	190-32-142-09114-01	0	12/15/2007	190-32-142-09114-01	
			<i>To correct software issue during Pressure Pulse Testing.</i>					
			SB	190-32-142-09114-02	1	8/1/2022	190-32-142-09114-02	
			<i>To eliminate noise on the ALE line that causes intermittent microcontroller resets.</i>					
			SB	190-32-142-09114-03	0	2/8/2023	190-32-142-09114-03	
			<i>(Description not available)</i>					
			SB	190-32-142-091141-01	0	11/11/2009	190-32-142-091141-01	
			<i>To update the BCM SW to correct the manual autobrake disarm and to remove the ac Family Comparison Test.</i>					
			SB	190-32-142-091142-01	0	10/1/2014	190-32-142-091142-01	
			<i>BCM software modification to P/N 142-091143.</i>					
			SB	190-32-142-091143-01	0	7/17/2017	190-32-142-091143-01	
			<i>This modification will eliminate noise on the ALE line that cause intermittent reset/transient BRK LH FAULT and BRK RH FAULT messages to the Embraer 170/190 Engine Indicating Crew Alert System (EICAS).</i>					
142-093	HYTROL MARK V BRAKE CONTROL UNIT (BCU)	EMB-145, 145-AEW&C	CMM	32-43-79	2	6/30/2016	CMM142-093	
142-101	BRAKE SYSTEM CONTROL UNIT (BSCU)	B-777-300	CMM	32-45-10	2	3/15/2007	CMM142-101	
			ATLAS	N/A	-	12/1/2003	AT142101	
142-103	DIGITAL MARK IV ANTISKID CONTROL UNIT (DACU)	CESSNA 525B CJ3	CMM	32-45-34	0	3/10/2005	CMM142-103	
142-107	HYTROL MARK III SKID CONTROL UNIT (SCU)	C-27J	CMM	32-45-54	0	1/1/2016	CMM142-107	
142-109	ANTISKID CONTROL UNIT (ACU)	MD-10	CMM	32-45-15	2	8/15/2019	CMM142-109	

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142-123, -01	BRAKE SYSTEM CONTROL UNIT (BSCU)	B-777-200	CMM	32-45-22	7	4/23/2020	CMM142-123		
			SB	142-123-32-1	0	6/21/2019	142-123-32-1		
			<i>provides instructions to replace obsolete microcontrollers and convert BSCUs, P/N 142-123 to P/N 142-123-01</i>						
			TSDP	N/A	C	9/6/2016	TSDP-142-123		
			TS	N/A	B	9/1/2009	TRD-142-123		
			TS	N/A	0	11/3/2015	TRD-142-123-01		
142-127	TBD	EMB 314 (SUPER TACANO)	CMM	32-43-81	0	5/14/2010	CMM142-127		
142-139-05, -06	EMERGENCY BRAKE CONTROL UNIT (EBCU)	A400M	CMP	32-45-31	003	10/25/2019	CMP142-139		
142-147	ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	737-6/7/900	CMM	32-46-01	4	3/30/2023	CMM142-147		
			SB	142-147-32-1	0	1/24/2012	142-147-32-1		
			<i>Provides instructions to return the AACUs to Crane Aerospace &amp; Electronics, Hydro-Aire, Inc. for transistor replacement on attrition basis only.</i>						
			SIL	SIL142-147-2	0	8/28/2012	SIL142-147-2		
			TRD	TRD142-147	C	11/10/2015	TRD142-147		
			TSDP	TSDP-142-147	B	5/7/2019	TSDP-142-147		
142-153, -1	DIGITAL MARK IV SCBS CONTROL UNIT	CESSNA	CMM	32-46-11	0	12/12/2018	142-153		
142-157	DIGITAL MARK IV SCBS CONTROL BOX	HA-420	CMM	32-46-55	0	5/30/2021	CMM142-157		
142-159	BRAKE SYSTEM CONTROL UNIT (BSCU)	747-8	CMM	32-45-75	3	7/15/2020	CMM142-159		
			TSDP	N/A	B	3/27/2011	TSDP142-159		

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142-177	BRAKE CONTROL & MONITORING UNIT (BCMU)	C919	CMM	32-41-03	1	11/1/2023	CMM142-177	R
142-199	ALTERNATE BRAKE CONTROL UNIT (ABCU)	C919	CMM	32-42-01	1	11/1/2023	CMM142-199	R
142-201	DIGITAL MARK IV SCBS CONTROL BOX	PC-24	CMM	32-41-80	1	10/3/2023	CMM142-201	R
			SB	142-201-32-1	0	10/21/2019	142-201-32-1	
			<i>Upgrade of Digital Antiskid Control System (DACU) p/n 142-201 to p/n 142-201-01</i>					
142-20714-2	BRAKE CONTROL MODULE (BCM)	E2	CMM	32-41-44	1	9/28/2021	CMM142-20714	
			SB	190E2-32-142-20714-2-01	0	7/28/2021	190E2-32-142-20714-2-01	
			<i>Modification to upgrade Brake Control Module (BCM) Assembly, pn 142-20714-2</i>					
			SB	190E2-32-142-20714-2-02	0	5/10/2023	190E2-32-142-20714-2-02	R
			<i>(Description not available)</i>					
142-219, 142-219-1	ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	737MAX, 737MAX10	CMM	32-46-15	3	5/7/2019	CMM142-219	
			TRD	N/A	D	7/11/2019	TRD-142-219	
			TRD	N/A	A	7/11/2019	TRD-142-219-1	
			TSDP	N/A	A	7/11/2019	TSDP-142-219	
142-225	BRAKE CONTROL UNIT (BCU)	CESSNA 700	CMM	32-46-20	0	10/19/2020	CMM142-225	
			SB	142-225-32-1	0	9/26/2023	142-225-32-1	R
			<i>(Description not available)</i>					
142-241	TBD	Project 1032	CMM	32-46-50	0	4/14/2019	CMM142-241	
160-011, 160-065	FUEL BOOSTER PUMP	DHC-7	OHM	28-20-28	3	4/1/1986	OHM160-011	
160-033	FUEL BOOST INVERMOTOR PUMP	EMB-110	CMM	28-20-30	1	4/30/1991	CMM160-033	

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160-093-1	TBD	CL41A	CMM	28-20-31	0	3/25/1982	CMM160-093-1
160-133A	SECONDARY ELECTRICAL (FUEL) BOOST PUMP ASSEMBLY	CESSNA 650	CMM	28-20-33	0	10/10/1983	CMM160-133A
160-151-5,-6; 160-15158; 160-15106; 160-15108	FUEL BOOST CANISTER PUMP ASSEMBLY	CANADAIR CL600	CMM	28-20-32	1	5/4/2005	CMM160-151-5
20-109500-1	BRAKE SYSTEM — RETURN PEDAL TRANSDUCER ASSEMBLY (PTA)	BD-100	Ref. Supplier CMM	(No ATA Number)	0	1/1/1901	N/A
			SB	20-109500-1-32-01	0	5/7/2014	20-109500-1-32-01
<i>To rework all effected LVDTs that have been found to be out of compliance with corrosion requirements per Crane A&amp;E SCD. The rework will replace the rod end with a new rod end that meets requirements.</i>							
20-171500, 20-171501	PEDAL TRANSDUCER ASSEMBLY (PTA)	ERJ-170/-190	CMM	32-41-01	3	10/1/2016	CMM20-171500
299-005	TEST SET ASSEMBLY	B-747-400	OMM	(No ATA Number)	2	2/15/2013	OMM299-005
			SB	299-005-32-197	0	3/10/1992	299-005-32-197
			<i>To allow test set to test control units, P/N 42-747-2 and 42-747-3.</i>				
			SB	299-005-32-2	1	5/12/2000	299-005-32-2
<i>Upgrade test set by replacing U3 with EPROM U4.</i>							
299-043	TEST SET P/N 299-043	B-737-700/737-800	O&M	(No ATA Number)	0	11/30/1999	O&M299-043
30-107	MANUAL OVERRIDE MOTOR OPERATED GATE VALVE	B-707	OHM	38-02-01	0	3/15/1958	OHM30-107
314160, -2	HYDRAULIT RESERVOIR PRESSURE REGULATORS	B-707/727/737	OHM	29-01-02	7	5/15/1978	OHM314160
314205, -3; 314900, -3	HYDRAULIC RELIEF VALVE	DC-8	OHM	29-10-1	0	3/30/1964	OHM314205
316525	HYDRAULIC HAND PUMP	B-707/727, DC-9/MD-80	OHM	29-20-46	1	8/10/1979	OHM316525

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316565	HYDRAULIC RELIEF VALVE ASSEMBLY	B-707/727/737/737-700/800	CMM	29-20-18	1	11/21/2018	316565
			SB	316565-32-144	0	10/1/1985	316565-32-144
			<i>Addition of flow arrows more clearly identify flow direction.</i>				
32-023, -1, -2	POWER BRAKE METERING CONTROL VALVE ASSEMBLY	CESSNA 650	CMM	32-42-79	1	8/15/2008	CMM32-023
			SB	32-023-32-149	0	4/10/1986	32-023-32-149
			<i>Replaces piston and rod assemblies P/N 32-02326 in power brake metering control valve with 32-02337</i>				
32-051	POWER BRAKE VALVE	BAE JETSTREAM 41	CMM	32-43-32	0	6/15/1991	CMM32-051
32-051-1	POWER BRAKE METERING VALVE	BAE JETSTREAM 41	CMM	32-43-44	0	5/15/1994	CMM32-051-1
32-077, -1	BRAKE METERING VALVE (BMV) ASSEMBLY	CESSNA 560 EXCEL	CMM	32-41-25	0	9/25/2006	CMM32-077
			TR	TR32-41-25-01	0	1/4/2018	TR32-41-25-01
			<i>Revise ASSEMBLY, Page 7002.</i>				
			TR	TR32-41-25-02	0	1/4/2018	TR32-41-25-02
			<i>Revises ASSEMBLY, Page 7012.</i>				
			TR	TR32-41-25-03	0	1/4/2018	TR32-41-25-03
			<i>Revises SPECIAL TOOLS, FIXTURES AND EQUIPMENT Page 9003</i>				
			TR	TR32-41-25-04	0	9/30/2019	TR32-41-25-04
<i>Revises REPAIR, Page 6003 and 6005.</i>							
TR	TR32-41-25-05	0	10/1/2019	TR32-41-25-05			
<i>Revises REPAIR, Page 9004.</i>							
TR	TR32-41-25-06	0	10/1/2019	TR32-41-25-06			
<i>Revises ILLUSTRATED PARTS LIST, Page 10014.</i>							
TR	TR32-41-25-07	0	10/2/2019	TR32-41-25-07			
<i>Revises ILLUSTRATED PARTS LIST, Page 10018.</i>							

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32-079	BRAKE METERING VALVE (BMV)	CESSNA 560	CMM	32-41-22	0	2/28/2008	CMM32-079
			TR	TR32-41-22-01	0	10/6/2020	TR32-41-22-01
			<i>Revises CMM 32-41-22, Initial Issue, Page 7004, Figure 7001 (Sheet 1 of 4).</i>				
			TR	TR32-41-22-02	0	10/6/2020	TR32-41-22-02
			<i>(Description not available)</i>				
			TR	TR32-41-22-03	0	10/6/2020	TR32-41-22-03
			<i>(Description not available)</i>				
32-089, -1, -2	BRAKE METERING VALVE (BMV) ASSEMBLY	CESSNA CITATION 680	CMM	32-41-27	0	8/20/2004	CMM32-089
32-105, -1, -2, -3, -4	BRAKE METERING VALVE (BMV)	CESSNA 560 EXCEL (AD00-5)	CMM	32-41-31	-	2/17/2023	CMM32-105
32-107-1	BRAKE METERING VALVE (BMV)	Cessna 525B, ADD99-8	CMM	32-48-51	1	5/13/2022	CMM32-107
32-109	BRAKE METERING VALVE (BMV)	CESSNA CJ2	CMM	32-48-47	0	10/13/2020	CMM32-109
32-117	BRAKE METERING VALVE (BMV)□	CESSNA CJ4	CMM	32-45-76	0	6/6/2020	CMM32-117
33-019, -1, 33-021	LANDING MANIFOLD ASSEMBLY	DC-9AB, DC-10AB, MD-80AB, B-717	CMM	32-42-80	4	3/31/2000	CMM33-019
			SB	33-019_021-32-87	1	10/9/1985	33-019_021-32-87
			<i>Replace NAS1611-112 packing used in 38-559 shuttle valve assembly with packing P/N 2-112N.</i>				
			TR	TR32-42-80-1	0	8/29/2014	TR32-42-80-1
			<i>Revises Page 701, Table 701</i>				
			TR	TR32-42-80-2	0	8/29/2014	TR32-42-80-2
			<i>Revises Page 705, Table 701.</i>				
33-065	TBD	DC-9	MAINT	(No ATA Number)	0	2/1/1965	MAINT33-065

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.	
33-065-1, -2; 33-067-1, -2	NORMAL ANTISKID VALVE MODULE (AVM)	B-747/757/767	CMM	32-42-51	14	6/14/2023	CMM33-065	R
			SB	33-065-1-32-193	1	8/5/2005	33-065-1-32-193	
			<i>To rework servovalve body to eliminate possibility of splitting with subsequent loss of hydraulic fluid.</i>					
			SB	33-065-1-32-2	0	5/22/1998	33-065-1-32-2	
			<i>Modifies 33-065-1 to 33-065-2.</i>					
			SB	33-067-1-32-156	1	2/16/1989	33-067-1-32-156	
			<i>To convert antiskid module P/N 33-067 to "unigain" operation compatible with carbon brakes.</i>					
33-073, -1	HYDRAULIC ACCUMULATOR ASSEMBLY	BEECHJET 400T (TBD)	CMM	32-45-24	0	10/22/2004	CMM33-073	
33-109-1, 33-109101-1, 33-111-1, 33-111101-1	ANTISKID VALVE MODULE	B-777	CMM	32-43-40	3	5/31/2022	CMM33-109_33-111	
33-117500	BRAKE SYSTEM	Global Express	SB	33-117500-32-01	0	2/2/2012	33-117500-32-01	
			<i>Provides instructions to inspect and, if required, replace the accumulator.</i>					
33-119500, -2	75 CUBIC INCH HYDRAULIC ACCUMULATOR	Global Express	SB	33-119500-32-01	0	2/12/2012	33-119500-32-01	
			<i>Provides instructions to inspect and, if required, replace the accumulator.</i>					
			SB	33-119500-32-02	1	12/17/2014	33-119500-32-02	
			<i>Crane A&amp;E, Hydro-Aire, Inc. has developed a new manifold that addresses Nitrogen leakage to make a new accumulator configuration P/N 33-119500-2.</i>					
33-147500	ACCUMULATOR	BD100	SB	33-147500-32-01	2	1/25/2012	33-147500-32-01	
			<i>(Description not available)</i>					
33-155500	ACCUMULATOR	BD100	SB	33-155500-32-01	2	1/25/2012	33-155500-32-01	
			<i>(Description not available)</i>					
33-159500-1	BRAKE ACCUMULATOR ASSEMBLY	ERJ-170/-190	CMM (Supsd by Triumph CMM 32-44- 13_R0)	32-44-13	2	10/1/2015	CMM33-159500-1	

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33-175	CENTER LANDING GEAR (CLG) ANTISKID MANIFOLD ASSEMBLY	MD-10	CMM	32-45-11	1	7/1/2017	CMM33-175
33-177, 33-177-1	MAIN LANDING GEAR (MLG) ANTISKID MANIFOLD ASSEMBLY	MD-10	CMM	32-45-12	3	7/28/2017	CMM33-177
			SB	33-177-32-1	0	3/8/2007	33-177-32-1
			<i>To prevent wear and lockup of the valve lap assembly in the pressure to brake position or brake to return position.</i>				
			SB	33-177-32-2	0	8/30/2007	33-177-32-2
			<i>Converts the MLG Antiskid Manifold Assembly from a Multigain operation to a Unigain operation. Change assists the brake system with the carbon "grabby brake" complaints.</i>				
35-055	TBD	L-188	CMM	75-02-02	2	7/3/2008	CMM35-055
35-381A	ANTI-ICING SOLENOID VALVE	GENERAL ELECTRIC	OHM	75-10-3	0	12/15/1960	OHM35-381A
35-813	HYDRAULIC RESERVOIR AIR PRESSURE REGULATOR CARTRIDGE VALVE	BOEING HYDROFOIL	OHM	29-20-63	0	5/15/1974	OHM35-813
35-833500	REMOTE CHARGE VALVE (RCV)	ERJ-170	CMM	32-44-15	3	10/1/2016	CMM35-833500
35-847500	REMOTE GAUGE AND CHARGING VALVE	C919	ACMM	32-43-03	0	8/30/2019	ACMM35-847500
35-851500	REMOTE CHARGE VALVE (RCV)	E2	CMM	32-44-64	1	6/8/2023	CMM35-851500 R
37-001	SERVO VALVE ASSEMBLY	L-382C	OHM	32-40-62	0	3/17/1980	OHM37-001
37-015	4-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-48-01	1	3/1/2012	CMM37-015

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37-019, -1	4-WAY SERVOVALVE ASSEMBLY	B-707/727 MKII, DC-8	CMM	32-40-63	0	12/15/2011	CMM37-019
			SB	37-019-32-45	1	9/1/1979	37-019-32-45
			<i>Incorporates an improved cover assembly to allow water removal and provides a greater thermal stability with much less maintenance required.</i>				
37-123	LOAD COMPRESSOR CONTROLLER ASSEMBLY	L-1011	CMM	49-51-03	0	1/2/2009	CMM37-123
37-125	FREE TURBINE SPEED CONTROLLER ASSEMBLY	L-1011	CMM	49-31-05	0	1/1/2009	CMM37-125
37-161, -1	SERVOVALVE ASSEMBLY	B-727 MKIII, B-737	SB	37-161-32-63	0	3/28/1979	37-161-32-63
			<i>Improved means of retaining magnet and improved thermal stability.</i>				
37-161, 37-161-1	3-WAY SERVOVALVE ASSEMBLY	VARIOUS	CMM	32-40-64	0	3/1/2013	CMM37-161
37-195	HYTROL MARK III THREE-WAY SERVO VALVE ASSEMBLY	DHC-6	OHM	32-42-14	0	5/15/1973	OHM37-195
37-213	4-WAY SERVOVALVE ASSEMBLY	B-707/727 MKII	CMM	32-40-71	1	12/10/2011	CMM37-213
37-231	3-WAY SERVOVALVE ASSEMBLY	B-727 MKIII, B-737	CMM	32-40-75	8	7/1/2015	CMM37-231
37-245	4-WAY SERVOVALVE ASSEMBLY	DC-9-10/20/30/40	CMM	32-42-46	3	11/18/2011	CMM37-245
37-259	3-WAY SERVOVALVE ASSEMBLY	DC-9-50/80, MD-80	OHM	32-42-34	4	3/1/2012	CMM37-259
37-285	4-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-48-02	0	11/25/2010	CMM37-285
37-295	3-WAY SERVOVALVE ASSEMBLY	VARIOUS	CMM	32-48-07	0	3/1/2012	CMM37-295

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37-305	3-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-43-97	0	5/14/2010	CMM37-305
37-331	3-WAY SERVOVALVE ASSEMBLY	VARIOUS	CMM	32-48-37	0	7/15/2011	CMM37-331
37-333	3-WAY SERVOVALVE ASSEMBLY	Saab 340	CMM	32-48-04	0	11/20/2010	CMM37-333
37-337	3-WAY SERVOVALVE ASSEMBLY	DHC-8	CMM	32-48-23	0	7/1/2014	CMM37-337
37-339	3-WAY SERVOVALVE ASSEMBLY	ATR-42	CMM	32-48-21	0	5/1/2011	CMM37-339
37-341, -1, -2	3-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-48-06	0	12/1/2010	CMM37-341
37-355	3-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-43-94	1	8/10/2011	CMM37-355
37-371	3-WAY SERVOVALVE ASSEMBLY	ATR 72	CMM	32-43-95	0	8/10/2011	CMM37-371
37-389	3-WAY SERVOVALVE ASSEMBLY	VARIOUS	CMM	32-48-25	0	5/31/2020	CMM37-389
37-391	3-WAY SERVOVALVE ASSEMBLY	CL601	CMM	32-43-70	2	7/15/2011	CMM37-391
37-397	3-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-48-05	0	12/15/2010	CMM37-397
37-399	3-WAY SERVOVALVE ASSEMBLY	Saab 2000	CMM	32-43-96	0	8/15/2007	CMM37-399
37-409, -2, -3	3-WAY SERVOVALVE ASSEMBLY	ERJ190	CMM	32-48-33	1	8/28/2010	CMM37-409

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37-411	3-WAY SERVOVALVE ASSEMBLY	Various	CMM	32-45-63	0	5/15/2008	CMM37-411
37-413	TBD	Various (Mil)	CMM	32-48-24	0	12/1/2010	CMM37-413
37-415	3-WAY SERVOVALVE ASSEMBLY	B-717	CMM	32-48-48	0	2/15/2013	CMM37-415
37-421	3-WAY SERVOVALVE ASSEMBLY	560 EXCEL	CMM	32-48-31	1	6/12/2010	CMM37-421
37-433, -1	3-WAY SERVOVALVE ASSEMBLY	ERJ-170	CMM	32-45-59	2	3/1/2013	CMM37-433
37-435	3-WAY SERVOVALVE ASSEMBLY	Aermacchi M346	CMM	32-48-94	0	2/1/2013	CMM37-435
37-441, -1	3-WAY SERVOVALVE ASSEMBLY	AD00-5	CMM	32-48-55	0	12/15/2011	CMM37-441
37-509	3-WAY SERVOVALVE ASSEMBLY	Cessna 525	CMM	32-48-41	0	12/15/2011	CMM37-509
37-515	3-WAY SERVOVALVE ASSEMBLY	CL-604	CMM	32-48-46	0	4/18/2019	CMM37-515
37-517	3-WAY SERVOVALVE ASSEMBLY	EMB -145	CMM	32-45-58	2	2/15/2013	CMM37-517
37-525	3-WAY SERVOVALVE ASSEMBLY	GE KL-1	CMM	32-48-90	0	1/15/2013	CMM37-525
37-531	3-WAY SERVOVALVE ASSEMBLY	Cessna CJ2	CMM	32-48-45	0	8/23/2010	CMM37-531
37-539	3-WAY SERVOVALVE ASSEMBLY	HONDAJET	CMM	32-48-91	0	6/5/2018	CMM37-539

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37-979-1, 37-979-2	3-WAY SERVOVALVE ASSEMBLY	737 MAX	CMM	32-48-96	0	9/8/2017	CMM37-979
			TR	TR32-48-96-01	0	2/28/2018	TR32-48-96-01
			<i>(Description not available)</i>				
			TR	TR32-48-96-02	0	2/28/2018	TR32-48-96-02
			<i>(Description not available)</i>				
			TR	TR32-48-96-03	0	2/28/2018	TR32-48-96-03
			<i>(Description not available)</i>				
			TR	TR32-48-96-04	0	2/28/2018	TR32-48-96-04
			<i>(Description not available)</i>				
			TR	TR32-48-96-05	0	2/28/2018	TR32-48-96-05
<i>(Description not available)</i>							
TR	TR32-48-96-06	0	2/28/2018	TR32-48-96-06			
<i>(Description not available)</i>							
TR	TR32-48-96-07	0	2/28/2018	TR32-48-96-07			
<i>(Description not available)</i>							
TR	TR32-48-96-08	0	2/28/2018	TR32-48-96-08			
<i>(Description not available)</i>							
TR	TR32-48-96-09	0	2/28/2018	TR32-48-96-09			
<i>(Description not available)</i>							
38-113C	HYTROL ANTI-SKID BY-PASS PRESSURE MODULATOR	CV-800	OHM	32-40-0	0	8/1/1960	OHM38-113C
38-129D	0.75 INCH BY-PASS RESTRICTOR VALVE ASSEMBLY	CV-800	OHM	32-40-00	0	10/1/1961	OHM38-129D
38-131A	DUAL ANTI-SKID PILOT OPERATED VALVE	DC-8	OHM	32-40-0	2	6/20/1968	OHM38-131A
38-133 SERIES	PRESSURE MODULATOR	YS-11	OHM	32-40-32	0	7/15/1971	OHM38-133

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38-269A	HYTROL ANTI-SKID PRESSURE MODULATOR	CV-880	OHM	32-40-0	0	5/1/1960	OHM38-269A
38-289	3000 PSI SOLENOID OPERATED VALVE	CV-880	OHM	32-40-0	0	5/1/1960	OHM38-289
38-299	DUAL ANTI-SKID PILOT OPERATED VALVE	CV-880	OHM	32-40-0	0	5/15/1964	OHM38-299
38-367,-1	HYTROL ANTI-SKID PRESSURE MODULATOR	CV-600	OHM	32-40-0	0	2/1/1967	OHM38-367
38-493	PRESSURE RELIEF CARTRIDGE VALVE	BOEING HYDROFOIL	OHM	32-40-70	0	8/1/1974	OHM38-493
38-497	HYDRAULIC RELIEF VALVE ASSEMBLY	BOEING HYDROFOIL	OHM	(No ATA Number)	0	7/1/1974	OHM38-497
38-529	POWER BRAKE RELAY VALVE (PBRV)	CESSNA 500	CMM	32-40-94	1	6/30/2011	CMM38-529
38-535	POWER BRAKE RELAY VALVE	METROLINER	CMM	32-40-89	0	11/18/1980	CMM38-535
38-579	THREE-WAY SELECTOR VALVE	B-747	OHM	32-40-81	2	8/26/2004	CMM38-579
38-581-3, -4	POWER BRAKE RELAY VALVE (PBRV)	Various	CMM	32-40-92	1	8/23/2010	CMM38-581-3
38-675,-1	POWER BRAKE RELAY VALVE (PBRV)	BJET400	CMM	32-40-96	3	2/18/2011	CMM38-675
38-691, -1, -2	POWER BRAKE RELAY VALVE (PBRV)	Beech 1900	CMM	32-48-03	0	4/20/2008	CMM38-691
38-693	POWER BRAKE RELAY VALVE (PBRV)	CESSNA 550ADV	CMM	32-40-97	1	10/21/2011	CMM38-693

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
38-707-1, -2, -3	POWER BRAKE RELAY VALVE (PBRV)	CESSNA 550	CMM	32-40-99	2	10/16/2006	CMM38-707-1
38-747	POWER BRAKE RELAY VALVE (PBRV)	CESSNA 525	CMM	32-43-42	1	12/15/2011	CMM38-747
38-771	POWER BRAKE RELAY VALVE (PBVR)	BEECHJET400/400A	CMM	32-43-27	2	9/27/2011	CMM38-771
38-793500, -1	3-WAY SHUTOFF VALVE	EMB 145	SL <i>(Description not available)</i>	38-793500-1	0	5/26/2004	38-793500-1
38-797500	3-WAY SHUTOFF VALVE (SOV)	Bombardier BD700	SB <i>(Description not available)</i>	SB 38-797500-32-02	2	11/30/2023	SB 38-797500-32-02 R

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
38-805, 38-805-2	GROUND SPOILER INTERLOCK VALVE ASSEMBLY	B-737-600/700/800	CMM	27-50-01	0	6/15/2011	CMM38-805
			SB	38-805-27-A	0	3/9/2009	38-805-27-A
				<i>Modification of Ground Spoiler Interlock Valve from P/N 38-805 to 38-805-A.</i>			
			SB	38-805-32-1	0	5/30/2000	38-805-32-1
				<i>Installation of anti-chafe washers; repair of anodic coating</i>			
			SB	SB38-805-27-2	0	8/9/2010	SB38-805-27-2
				<i>Modification of Ground Spoiler Interlock Valve from P/N 38-805 to 38-805-2.</i>			
			TR	TR27-50-01-01	0	3/15/2016	TR27-50-01-01
				<i>Change NULL POSITION VERIFICATION, Step 4</i>			
			TR	TR27-50-01-02	0	9/12/2016	TR27-50-01-02
				<i>Change IPL Figure 1, Item 90</i>			
			TR	TR27-50-01-03	0	9/9/2016	TR27-50-01-03
				<i>Replace Figure 7001, Sheet 1 of 4</i>			
			TR	TR27-50-01-04	0	3/7/2017	TR27-50-01-04
				<i>Change Step (4)(f) and Step (4)(h)</i>			
			TR	TR27-50-01-05	0	3/7/2017	TR27-50-01-05
				<i>Change Step (4)(b) and Step (6)</i>			
			TR	TR27-50-01-06	0	3/7/2017	TR27-50-01-06
				<i>Change Step 2.D</i>			
			TR	TR27-50-01-07	0	4/16/2018	TR27-50-01-07
				<i>Replace IPL Figure 1 (Sheet 1 of 5).</i>			
			TR	TR27-50-01-08	0	4/27/2018	TR27-50-01-08
				<i>Change IPL Figure 1, Item 100</i>			
			TR	TR27-50-01-09	0	2/7/2020	TR27-50-01-09
				<i>Add Cor-Ban 27L (BMS 3-38) (Alternate to BMS 3-27) to Table 6002.</i>			
			TR	TR27-50-01-10	0	2/7/2020	TR27-50-01-10
				<i>Add Cor-Ban 27L (BMS 3-38) (Alternate to BMS 3-27) to Table 7002.</i>			
			TR	TR27-50-01-11	0	2/7/2020	TR27-50-01-11
				<i>Add Cor-Ban 27L (BMS 3-38) (Alternate to BMS 3-27) to Table 9002.</i>			

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38-805, 38-805-2	GROUND SPOILER INTERLOCK VALVE ASSEMBLY	B-737-600/700/800	TR	TR27-50-01-12	0	2/7/2020	TR27-50-01-12
<i>Add Zip-Chem Products CAGE Code V6KCJ9</i>							
38-811, -1	POWER BRAKE AND ANTISKID CONTROL VALVE	PREMIER I	CMM	32-43-72	1	3/15/2006	CMM38-811
38-861, -1	POWER BRAKE AND ANTISKID CONTROL VALVE	Cessna 525A (CJ2)	CMM	32-43-03	1	6/7/2002	CMM38-861
38-877500	BRAKE SYSTEM	Global Express	SB	38-877500-1-32-01	1	1/24/2011	38-877500-1-32-01
			<i>To retorque endcap on 3-Way Shutoff Valve to ensure the internal pressure does not affect end cap torque.</i>				
			SB	38-877500-2-32-01	0	1/24/2011	38-877500-2-32-01
<i>(Description not available)</i>							
38-907501, 38-907501-1, 38-907501-2	DUAL EMERGENCY/PARKING BRAKE VALVE ASSEMBLY	ERJ-170/-190	CMM	32-44-01	4	3/10/2022	CMM38-907501
			SB	170-32-38-907501-1-02	1	8/10/2018	170-32-38-907501-1-02
			<i>To solve leakage problems that occurred because of a loose adjustment screw on the Thermal Relief Valve (TRV) assembly.</i>				
			SB	170-32-38-907501-2-01	0	12/7/2018	170-32-38-907501-2-01
<i>This bulletin gives instructions to install the new TRV assemblies P/N 32-085120-3 (replaces the current TRV P/N 32-085120-1).</i>							
38-909500, -1, -2, -3, -4	BRAKE HYDRAULIC FUSE AND PARKING HYDRAULIC FUSE	ERJ-170/-190	CMM	32-41-15	3	10/1/2015	CMM38-909500
38-919500	CHECK VALVE	ERJ-170/-190	CMM	32-41-17	3	10/1/2016	CMM38-919500
38-947500, -1	3-WAY, 2-POSITION, SOLENOID-OPERATED SHUTOFF VALVE (SOV)	ERJ-170/-190	CMM	32-41-03	3	10/1/2015	CMM38-947500
38-949	MANUAL REVERSION VALVE ASSEMBLY	Pilatus PC-21	CMM	32-41-74	0	10/11/2022	CMM38-949

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
38-957500, 38-957500-1, 38-957500-2	3-WAY, 2-POSITION, SOLENOID-OPERATED SHUTOFF VALVE (SOV)	ERJ-190	CMM	32-41-29	4	10/1/2015	CMM38-957500		
			SB	190-32-38-957500-1-01	2	7/31/2008	190-32-38-957500-1-01		
			<i>To change P/N 38-957500 to -1 to improve brake system and SOV reliability.</i>						
			SB	190-32-38-957500-2-01	1	8/9/2011	190-32-38-957500-2-01		
<i>To change PN 38-957500-1 SOVs to PN 38-957500-2. Replaces the aluminum end cap P/N 36780 with a steel endcap P/N 37543 with a higher tightening torque.</i>									
39-023, -1	TBD	?	?	32-42-86	0	1/1/1901	MANUAL39-023		
39-023A, 39-023C	DUAL PRESSURE CONTROL VALVE	CV-990	OHM	32-40-0	0	4/15/1966	OHM39-023		
			SB	8-118-1	0	3/15/1965	8-118-1		
<i>Improve performance (frequency response) of control valve.</i>									
39-027A/B	TBD	CV-990	OHM	32-40-0	1	11/25/1968	OHM39-027		
39-043A	DUAL PRESSURE CONTROL VALVE	B-727 MKII	OHM	32-40-23	4	7/15/1990	OHM39-043A		
			SB	32-3	0	8/21/1970	32-3		
			<i>Improve reliability of solenoid, position switch and shutoff valve.</i>						
			SB	39-043A-32-49	0	6/15/1978	39-043A-32-49		
			<i>Remove adjust park brake switch/solenoid components and replace with integrated solenoid and switch.</i>						
			SB	8-110-1	0	2/10/1967	8-110-1		
<i>Improve performance (frequency response) of control valve</i>									
			SB	8-110-3	1	8/15/1967	8-110-3		
<i>Improve assembly characteristics of backup rings of pilot servo ports.</i>									

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39-045	PRESSURE CONTROL VALVE	B-707	OHM	32-00-16	4	7/9/1980	OHM39-045		
			SB	39-045-8-115-1	0	3/1/1965	39-045-8-115-1		
								<i>Improve performance (frequency response) of control valve.</i>	
			SB	8-115-1A	0	8/10/1966	8-115-1A		
								<i>Prevent internal corrosion of connector (J1).</i>	
			SB	8-115-2	0	12/15/1966	8-115-2		
								<i>Improve operating characteristics of shutoff valve during parking.</i>	
39-045	PRESSURE CONTROL VALVE	B-707	SB	8-115-3	0	5/15/1967	8-115-3		
								<i>Improve assembly characteristics of backup rings in pilot servo ports.</i>	
			SL	39-045-8-115-2	0	12/10/1965	39-045-8-115-2		
								<i>(Description not available)</i>	
39-045	PRESSURE CONTROL VALVE	B-707	SL	39-045-8-115-3	0	8/17/1967	39-045-8-115-3		
								<i>(Description not available)</i>	
39-047	PRESSURE CONTROL VALVE	B-727	OHM	32-40-25	3	9/29/1980	OHM39-047		
			SB	8-116-1	1	8/15/1967	8-116-1		
					<i>Improve assembly characteristics of backup rings in pilot servo ports.</i>				



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.	
39-069A/B	DUAL PRESSURE CONTROL VALVES	DC-8-50/55/60	OHM	32-40-22	3	2/23/1981	OHM39-069	
			SB	32-4	0	10/1/1970	32-4	
								<i>Improve solenoid position switch operation.</i>
			SB	32-6	0	8/25/1971	32-6	
								<i>Replace solenoid to prevent moisture contamination.</i>
			SB	39-069B-32-4	0	10/1/1970	39-069B-32-4	
								<i>Provides information to replace MS24547-1 switch with 21SX39-T switch.</i>
			SB	39-069B-32-6	0	8/25/1971	39-069B-32-6	
								<i>Provides instructions to replace solenoid p/n 39-14337 with solenoid p/n 39-04379</i>
SB	39-069B-32-7	0	7/31/1998	39-069B-32-7				
					<i>Replacement of solenoid with one providing integrated partial indicator switch.</i>			
SB	8-126-1	0	7/15/1965	8-126-1				
					<i>Improve performance (frequency response) of control valve.</i>			
SB	8-126-2	2	9/15/1979	8-126-2				
					<i>Make valve compatible to MK III system</i>			



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.							
39-101, -1	DUAL PRESSURE CONTROL VALVE (PCV)	DC-9-10/20/30/40	CMM	32-43-02	5	2/1/2008	CMM39-101							
			SB	39-101-32-143	0	1/16/1987	39-101-32-143							
			<i>To provide an alternative receptacle which has an improved interfacial seal and gasket configuration and gold plated pins.</i>											
			SB	39-101-32-52	0	12/1/1978	39-101-32-52							
			<i>To reduce arcing across parking brake switch contacts.</i>											
			SB	8-105-1	0	5/20/1965	8-105-1							
			<i>Rework solenoid plunger and rocker arm to eliminate possibility of shutoff valve interference.</i>											
39-107	DUAL PRESSURE CONTROL VALVE	BAC 1-11	OHM	32-40-11	1	1/15/1980	OHM39-107							
								SB	8-105-2	0	9/20/1966	8-105-2		
								<i>Reduce external leakage and improve shutoff valve operation.</i>						
								SB	8-105-3	0	4/1/1968	8-105-3		
								<i>Replace relief valve package with plug.</i>						
39-149, -1	DUAL PRESSURE CONTROL VALVE (PCV)	JETSTAR-1, LEAR-25	CMM	32-42-28	0	3/1/2008	CMM39-149							
			SB	39-149-1-32-1	1	9/14/2010	39-149-1-32-1							
			<i>Noted during shop replacement of solenoid assemblies that moisture and contamination are present. Application of Pro-Seal 890B-2 sealant will extend the life of the PCV.</i>											
39-233-1A, -1B, -1C, 39-233-3A, -3B, -3C, -4, 39-491-1, -3, 39-525-1, -3	ANTISKID MODULAR ASSEMBLY	B-747-100, -200, -300	CMM	32-42-35	11	1/6/2016	CMM39-233-1A							
			SB	39-233-32-18	3	11/21/1980	39-233-32-18							
<i>Provide higher strength screws and improve corrosion resistance.</i>														



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39-239/A, B, C, 39-489, 39-521, 37-037A, B, C, 37-203	ANTISKID CONTROL VALVE (ACV) and SERVOVALVE ASSEMBLY	B-747	CMM	32-42-36	7	7/27/2007	CMM39-239	
			SB	32-16	2	10/19/1973	32-16	
							<i>Improve corrosion resistance.</i>	
			SB	37-037B-32-17	0	10/19/1973	37-037B-32-17	
							<i>Add null adjust screws, add spring pins and replace connector.</i>	
39-241	MARK III DUAL PRESSURE CONTROL VALVE	DC-8-63	OHM	32-40-21	1	8/30/1995	OHM39-241	
			SB	39-241-32-1	0	5/30/1998	39-241-32-1	
							<i>Removes switch and solenoid and replaces them with more reliable components.</i>	
39-245	DUAL PRESSURE CONTROL VALVE	C9B, DC-9-30	OHM	32-40-67	0	11/11/1981	OHM39-245	
			SB	39-245-32-203	1	3/1/1995	39-245-32-203	
							<i>Replace converters to prevent face seal deterioration.</i>	
			SB	39-245-32-53	0	12/1/1978	39-245-32-53	
				<i>To reduce arcing across parking brake switch contacts.</i>				
			SL	SIL39-101-001	1	2/21/2001	SIL39-101-001	
							<i>(Description not available)</i>	

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
39-249, 39-249-1, 39-249-2	DUAL PRESSURE CONTROL VALVE (PCV)	DC-9-50/MD-80	CMM	32-42-30	8	3/1/2008	CMM39-249		
			SB	39-249-1-32-194	2	7/17/1991	39-249-1-32-194		
								<i>To modify P/N 39-249 and 39-249-1 to 39-249-2 and to prevent leakage of parking brake shutoff valve.</i>	
			SB	39-249-32-101	0	5/25/1982	39-249-32-101		
								<i>39-36320-2 slide and sleeve assembly (material 440 CRES) is less susceptible to oxidation.</i>	
			SB	39-249-32-142	0	9/25/1985	39-249-32-142		
								<i>All four antiskid fault annunciators remain on following parking brake release.</i>	
			SB	39-249-32-169	0	11/17/1987	39-249-32-169		
					<i>Product improvement. Add packings to prevent water from entering the valve body slide and sleeve cavities causing corrosion.</i>				
39-309, 39-309-1, 39-309-1-3, 39-309-1-4, 39-309-2, 39-309-3, 39-309-3-3, 39-309-3-4, 39-309-4	DUAL PRESSURE CONTROL VALVE (PCV)	JETSTAR-2, LEAR-25,28,29,31,35,36,50,55,56,60	CMM	32-42-33	6	8/1/2015	CMM39-309		
			SB	39-309-2-32-1	1	9/14/2010	39-309-2-32-1		
								<i>Noted during shop replacement of solenoid assemblies that moisture and contamination are present. Application of Pro-Seal 890B-2 sealant will extend the life of the PCV.</i>	
			SB	39-309-2-32-2	0	4/14/2011	39-309-2-32-2		
								<i>Convert 39-309-1 to -3 and 39-309-2 to -4</i>	
39-333	HYTROL MARK III 3-WAY CONTROL VALVE ASSEMBLY	DHC-5, DHC-6	CMM	32-42-12	0	5/26/2012	CMM39-333		
39-353	CONTROL VALVE ASSEMBLY	B-727 MKIII, B-737	CMM	32-40-52	9	10/15/2016	CMM39-353		
39-463	BRAKE CONTROL VALVE	DHC-7	OHM	32-42-42	1	2/20/1986	OHM39-463		

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
39-475	DUAL PRESSURE CONTROL VALVE	DHC-7	OHM	32-40-68	0	4/15/1974	OHM39-475
39-483	DUAL PRESSURE CONTROL VALVE ASSEMBLY	B-727 MKII	OHM	32-40-69	5	3/15/2006	OHM39-483
39-537,-1	DUAL PRESSURE CONTROL VALVE	DHC-7	OHM	32-42-38	0	2/15/1979	OHM39-537
			SB	39-537-1-32-82	0	1/9/1981	39-537-1-32-82
			<i>To prevent possible fault due to nozzle flapper erosion.</i>				
39-537-2	DUAL PRESSURE CONTROL VALVE ASSEMBLY	DHC-7	CMM	32-43-39	0	7/31/1998	CMM39-537-2
39-557	CONTROL VALVE ASSEMBLY	DC-9AB, DC-10AB, MD-80AB	OHM	32-40-85	3	8/27/2004	OHM39-557
39-581, -1, -2, -3, -4	SKID CONTROL VALVE (SCV)	Various	CMM	32-45-62	1	6/15/2019	CMM39-581
39-613	ANTI SKID CONTROL VALVE	AERMACCHI MB339	CMM	32-40-43	0	6/17/1988	CMM39-613
39-617, 37-309	ANTISKID CONTROL VALVE (ACV) AND SERVOVALVE ASSEMBLY	B-757/767/747-400	CMM	32-42-53	11	4/3/2023	CMM39-617
			SB	39-617-32-1	1	11/11/1994	39-617-32-1
			<i>To prevent leakage at the filter retaining plug used on servovalve.</i>				
39-621-1	CONTROL VALVE ASSEMBLY	Various	CMM	32-43-98	0	7/31/2007	CMM39-621-1
39-647	CONTROL VALVE ASSEMBLY	SAAB 340	CMM	32-43-10	2	4/1/2008	CMM39-647
39-651-1, 39-651-2	POWER BRAKE AND ANTISKID CONTROL VALVE	EMB-120	CMM	32-42-92	4	2/15/2012	CMM39-651
			TR	TR32-42-92-01	0	7/30/2020	TR32-42-92-01
			<i>(Description not available)</i>				
			TR	TR32-42-92-02	0	7/30/2020	TR32-42-92-02
			<i>(Description not available)</i>				

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39-667	ANTISKID CONTROL VALVE (ACV)	CESSNA 650	CMM	32-42-61	3	4/20/2015	CMM39-667
39-671	CONTROL VALVE ASSEMBLY	DHC-8	CMM	32-42-76	4	4/12/2023	CMM39-671 R
			SB	39-671-32-1	0	10/18/2012	39-671-32-1
			<i>This SB provides a pressure test procedure to ensure the ofifice plug is not dislodged after exposure to 5000 psid (34475kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i>				
			TR	TR32-42-76-01	0	3/1/2017	TR32-42-76-01
			<i>(Description not available)</i>				
			TR	TR32-42-76-02	0	3/1/2017	TR32-42-76-02
			<i>(Description not available)</i>				
			TR	TR32-42-76-03	0	3/1/2017	TR32-42-76-03
			<i>(Description not available)</i>				
			TR	TR32-42-76-04	0	3/1/2017	TR32-42-76-04
			<i>(Description not available)</i>				
			TR	TR32-42-76-05	0	3/1/2017	TR32-42-76-05
			<i>(Description not available)</i>				
			TR	TR32-42-76-06	0	3/1/2017	TR32-42-76-06
			<i>(Description not available)</i>				

PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
39-685-1, -2, -3, -4	ANTISKID VALVE MANIFOLD (AVM)	ATR-42 Series	CMM	32-42-88	7	10/12/2018	CMM39-685		
			SB	39-685-32-1	0	1/9/2015	39-685-32-1		
			<i>This SB provides a pressure test procedure to ensure the ofifice plug is not dislodged after exposure to 5000 psid (34475kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i>						
			SB	39-685-32-210	0	7/20/1991	39-685-32-210		
			<i>Screw replacement with increased torque to prevent failure.</i>						
			SB	39-685-32-211	1	12/1/2005	39-685-32-211		
			<i>Product improvement to increase the breaking strength of the screws securing the shuttle valve covers.</i>						
			SB	39-685-3-32-13	0	3/30/2011	39-685-3-32-13		
<i>Add identification markings for the RHI, RHO, LHI, AND LHO electrical connectors and hydraulic ports.</i>									
			TR	TR32-42-88-01	0	3/18/2019	TR32-42-88-01		
<i>(Description not available)</i>									
			TR	TR32-42-88-02	0	3/29/2019	TR32-42-88-02		
<i>(Description not available)</i>									
39-723, 39-723-1	BRAKE FEEL AUGMENTER	VARIOUS	CMM	32-48-19	0	2/1/2013	CMM39-723		
39-729	CONTROL VALVE ASSEMBLY	B-757	CMM	32-42-02	2	1/31/2016	CMM39-729		

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39-739-1, -2, -3	HYDRAULIC ANTISKID VALVE MANIFOLD (AVM)	ATR-72	CMM	32-42-26	6	11/15/2016	CMM39-739		
			SB	39-739-1-32-12	0	3/30/2011	39-739-1-32-12		
			<i>To correct incorrect connector and port identification engravings with new identification labels.</i>						
			SB	39-739-2-32-01	0	3/5/2012	39-739-2-32-01		
			<i>Product improvement upgrade to increase the breaking strength of the screws securing the shuttle-valve covers.</i>						
			SB	39-739-32-1	0	1/9/2015	39-739-32-1		
			<i>This SB provides a pressure test procedure to ensure the office plug is not dislodged after exposure to 5000 psid (34475kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i>						
39-771	CONTROL VALVE ASSEMBLY	CRJ-100/200	CMM	32-43-29	3	8/15/2017	CMM39-771		
			SB	39-771-32-1	0	11/29/2012	39-771-32-1		
			<i>This SB provides a pressure test procedure to ensure the office plug is not dislodged after exposure to 5000 psid (34475kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i>						
			TR	TR32-42-26-01	0	3/18/2019	TR32-42-26-01		
<i>(Description not available)</i>									
39-773	CONTROL VALVE ASSEMBLY	BAE JETSTREAM 41, 31, 32	CMM	32-43-31	1	6/23/2008	CMM39-773		
			TR	TR32-42-26-02	0	3/18/2019	TR32-42-26-02		
<i>(Description not available)</i>									
39-777-1; 37-511-1	ANTISKID CONTROL VALVE (ACV), 3-WAY SERVOVALVE ASSEMBLY	B-777	CMM	32-43-43	7	8/31/2022	CMM39-777_37-511		
39-779	CONTROL VALVE ASSEMBLY	SAAB 2000	CMM	32-43-38	1	2/1/2008	CMM39-779		
39-783	CONTROL VALVE ASSEMBLY	CESSNA 750	CMM	32-43-55	2	9/19/2023	CMM39-783 R		

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39-787	CONTROL VALVE ASSEMBLY	LEAR 45	CMM	32-43-46	3	11/5/2019	CMM39-787
39-801	TBD	GULFSTREAM V	CMM	32-43-64	0	7/15/1995	CMM39-801
39-803	CONTROL VALVE ASSEMBLY	CL604	CMM	32-43-52	1	5/15/2017	CMM39-803
			SB	39-803-32-1	0	11/19/2012	39-803-32-1
<p><i>This SB provides a pressure test procedure to ensure the orifice plug is not dislodged after exposure to 5000 psid (34475 kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i></p>							
39-807	BRAKE CONTROL VALVE (BCV)	EMB-145, 145-AEW&C	CMM	32-41-06	2	8/15/2017	39-807
39-809	ANTISKID CONTROL VALVE (ACV)	GULFSTREAM V	CMM	32-45-02	2	10/16/2015	CMM39-809
39-815	TBD	GLOBAL EXPRESS	CMM	32-43-67	3	9/21/2009	CMM39-815
39-837, 39-837-1	CONTROL VALVE ASSEMBLY	CESSNA CITATION 560 ENCORE	CMM	32-41-23	2	12/12/2022	CMM39-837
39-853, -1	CONTROL VALVE ASSEMBLY	BD-100	CMM	32-43-11	4	9/1/2015	CMM39-853
			SB	39-853-1-32-1	1	1/18/2013	39-853-1-32-1
<p><i>This SB provides a pressure test procedure to ensure the orifice plug is not dislodged after exposure to 5000 psid (34475kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i></p>							
39-865, -1	ANTISKID CONTROL VALVE (ACV)	CESSNA CITATION 680 SOVEREIGN	CMM	32-41-28	2	10/18/2022	CMM39-865
			SB	39-865-32-1	0	2/1/2013	39-865-32-1
<p><i>This SB provides a pressure test procedure to ensure the orifice plug is not dislodged after exposure to 5000 psid (34475kPad) pressure applied in both directions. As an option the sleeve end assy can be replaced instead of performing pressure test.</i></p>							
39-869, 39-869-01	CONTROL VALVE ASSEMBLY	AERMACCHI M346	CMM	32-48-82	0	11/2/2020	CMM39-869

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
39-871,-1, -2	CONTROL VALVE ASSEMBLY	ERJ-170	CMM	32-41-09	6	2/8/2023	CMM39-871		
			SB	170-32-39-871-1-01	1	4/3/2009	170-32-39-871-1-01		
			<i>Modification to convert P/N 39-871-1 BCV to P/N 39-871-2 BCV to increase valve body fatigue strength, and to improve BCV stiffness to reduce the effects of vibration loads on BCV performance; Rev. 1 updates FOC expiration to 9/11/11.</i>						
			TR	TR32-41-09-01	0	11/2/2020	TR32-41-09-01		
			<i>Revises CMM 32-41-09, Revision 5, Pages 11, 13, 15, 1023, 7010, 7012, and 7014.</i>						
39-873	CONTROL VALVE ASSEMBLY	NORTHROP F-5	CMM	32-48-61	0	4/29/2021	CMM39-873		
			TR	TR32-41-09-02	0	11/23/2020	TR32-41-09-02		
			<i>Revises CMM 32-41-09, Revision 5, Page 10012.</i>						
39-877	CONTROL VALVE ASSEMBLY	GE KL-1	CMM	32-42-93	0	1/15/2013	CMM39-877		
39-887, 37-443	ANTISKID CONTROL VALVE (ACV), 3-WAY SERVOVALVE ASSEMBLY	MD-10	CMM	32-45-16	2	5/13/2020	CMM39-887		
39-891, -1, -2, -3	BRAKE CONTROL VALVE (BCV)	ERJ-190	CMM	32-41-20	7	12/15/2016	CMM39-891		
			SB	190-32-39-891-2-01	0	4/23/2007	190-32-39-891-2-01		
			<i>Modification to convert P/N 39-891 to P/N 39-891-2 and change P/N 39-891-1 to P/N 39-891-2.</i>						
39-907, 39-907-1	DUAL ANTISKID CONTROL VALVE (ACV) MANIFOLD ASSEMBLY	B-717	CMM	32-45-09	9	6/23/2022	CMM39-907		
			SB	39-907-32-1	2	1/29/2000	39-907-32-1		
<i>Design modification</i>									
39-913	ANTISKID CONTROL VALVE (ACV) ASSEMBLY	CESSNA 525B CJ3	CMM	32-45-36	1	3/14/2017	CMM39-913		

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39-919	ANTISKID CONTROL VALVE (ACV)	CESSNA CJ2	CMM	32-48-44	1	9/15/2020	CMM39-919
39-939	ANTISKID CONTROL VALVE (ACV)	CESSNA CJ4	CMM	32-48-54	0	6/15/2011	CMM39-939
39-967	TBD	C919	CMM	32-42-02	0	9/13/2019	CMM39-967
39-969, 39-969-01	DUAL BRAKE CONTROL VALVE (BCV)	C919	CMM	32-41-05	1	3/28/2022	CMM39-969
39-979	BRAKE CONTROL VALVE (BCV)	E2	CMM	32-41-46	0	8/10/2018	CMM39-979
39-987, 37-547	BRAKE CONTROL VALVE (BCV)	CESSNA 700	CMM	32-41-83	0	12/7/2020	CMM39-987
39-993-1	DUAL CHANNEL ANTISKID VALVE (ASV)	737 MAX	CMM	32-48-95	3	1/11/2019	CMM39-993
			SB	39-993-1-32-1	0	6/4/2018	39-993-1-32-1
			<i>Incorporation of this service bulletin replaces the existing guides and seats with revised guides and seats which accommodate the housing bore machining discrepancy and allow the outermost gland components to be fully supported.</i>				
40-219A-1, -2	TBD	DC-8	OHM	32-40-41	1	8/1/1971	OHM40-219A-1
40-221A	HYTROL ANTI-SKID CONTROL BOX	DC-8, YS-11	OHM	32-40-0	2	11/15/1968	OHM40-221A
			SB	32-4-64	0	3/15/1964	32-4-64
			<i>Increase service life to control box by replacing two secondary power relays having a higher operating range.</i>				
40-223A	HYTROL ANTI-SKID UNIT FOOT THUMPER	DC-8	OHM	(No ATA Number)	0	5/1/1959	OHM40-223A
40-229	HYTROL TEST BENCH	B-707, CV-880, DC-8, YS-11	OHM	32-40-0	0	11/15/1961	OHM40-229

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40-239	HYTROL SYSTEM FIELD TEST UNIT	B-707	OHM	32-40-0	0	2/1/1962	OHM40-239
40-249, A, B	ANTI-SKID CONTROL SHIELD	B-707	OHM	32-40-0	1	5/15/1966	OHM40-249
40-255A, B	HYTROL SKID DETECTOR	CV-880	OHM	32-40-0	1	8/15/1961	OHM40-255A
			SB	32-2-63	0	11/10/1963	32-2-63
			<i>Increase service life of detector.</i>				
40-255C	HYTROL SKID DETECTOR	CV-880	OHM	32-40-0	1	9/1/1969	OHM40-255C
40-263B	ANTI-SKID CONTROL BOX	CV-880	OHM	32-40-0	0	1/15/1961	OHM40-263B
40-273A	ANTI-SKID CONTROL BOX	CV-880	OHM	32-40-0	0	2/15/1961	OHM40-273A
			SB	32-40-0	0	6/1/1963	32-40-0
			<i>Replace shock-mount assembly and rearrange wire bundle.</i>				
40-293, A	HYTROL SKID DETECTOR ASSEMBLIES	B-707-120	OHM	32-40-0	1	1/15/1968	OHM40-293
40-303	HYTROL DETECTOR TEST BENCH	B-707, CV-880, DC-8, YS-11	OPER	32-40	0	2/15/1961	OPER40-303
40-355	HYTROL INSERT PANEL	B-707	OHM	32-40-0	0	2/1/1962	OHM40-355
40-371	HYTROL SYSTEM FIELD TEST UNIT	DC-8	OPER	32-40-0	0	4/15/1961	OPER40-371
40-373	HYTROL SYSTEM FIELD TEST UNIT	CV-880	OHM	32-40-0	0	2/1/1961	OHM40-373
40-375	HYTROL INSERT PANEL	CV-880	OHM	32-40-0	1	2/1/1962	OHM40-375
40-381	HYTROL INSERT PANEL	DC-8, YS-11	OHM	32-40-0	0	2/1/1962	OHM40-381
40-433B, 40-433C, 40-433L	WHEEL SPEED TRANSDUCER ASSEMBLY	CV-600,-990, LEAR-25,-35	OHM	32-40-02	0	7/28/1980	OHM40-433B

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40-457	ANTI-ICING RELAY ASSEMBLY	CV-880	OHM	30-00-0	0	2/1/1961	OHM40-457
40-493-1,-2	HYTROL SKID DETECTOR ASSEMBLY	B-707	OHM	32-40-0	3	8/1/1971	OHM40-493-1
40-501-1, -2	WHEEL SPEED DETECTOR ASSEMBLIES	YS-11	OHM	32-40-0	1	10/15/1965	OHM40-501-1
40-521	HYTROL CONTROL BOX	DC-8	OHM	32-40-0	0	2/1/1964	OHM40-521
40-533	WHEEL SPEED TRANSDUCER (WST)	B727	CMM	32-47-01	0	2/29/2008	CMM40-533
			SB	32-5-64	1	11/20/1969	32-5-64
			<i>(Description not available)</i>				
40-563, 40-563A, 40-563B	SKID CONTROL BOX	CV-990	OHM	32-40-0	0	2/20/1966	OHM40-563
40-569	HYTROL FIELD TESTER	DC-8F, DC-9, BAC 1-11, L-382C	MAINT	32-40-0	0	7/1/1963	MAINT40-569
40-575	HYTROL WHEEL SPEED TRANSDUCER ASSEMBLY	B-707	OHM	32-00-42	1	6/10/1971	OHM40-575
40-581	WHEEL SPEED TRANSDUCER (WST)	DC-8	OHM	32-40-11	3	10/11/2011	OHM40-581
40-621, -1	TBD	L-382C	OHM	32-40-09	1	10/4/1985	OHM40-621

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40-625, -2, -4, -4A, -5, -6, -7, -8	WHEEL SPEED AND SPOILER TRANSDUCER ASSEMBLY	DC-9, MD-80	CMM	32-42-03	8	4/30/2016	CMM40-625							
			SB	32-2 106-4	0	5/20/1970	8-106-4							
			<i>Prevent installation of locking in event transducer is not seated properly, and ensures that anti- rotation pin of adaptor is fully engaged in one of four slots of transducer.</i>											
			SB	40-625-5-32-3	0	8/21/2000	40-625-5-32-3							
			<i>Modifies the wheel speed transducer adapter to accept a new self-locking, stainless steel, set screw.</i>											
			SB	40-625-6-32-195	1	10/23/1992	40-625-6-32-195							
			<i>Change bearing configuration.</i>											
			SB	40-625-6-32-196	0	8/23/2000	40-625-6-32-196							
<i>Replaces the steel set screw with a self-locking, stainless steel set screw.</i>														
40-673-1	WHEEL SPEED TRANSDUCER	JETSTAR-1, JETSTAR-2	OHM	32-42-79	0	12/15/1975	OHM40-673-1							
								SB	8-106-1	0	5/15/1966	8-106-1		
								<i>Improve performance of transducer assembly.</i>						
								SB	8-106-2	0	8/1/1968	8-106-2		
								<i>Provides a more positive means of securing transducer adaptor.</i>						
40-787	TBD	B-SST	OHM	(No ATA Number)	0	6/15/1971	OHM40-787							
								SB	8-106-3	0	8/20/1969	8-106-3		
<i>Facilitate main landing gear wheel removal.</i>														
40-805, 40-80567, 40-80570	WHEEL SPEED TRANSDUCER (WST) and DRIVE COUPLING ASSEMBLY	B747	CMM	32-47-02	0	3/10/2008	CMM40-805							

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40-817	WHEEL SPEED TRANSDUCER (WST)	B-737	CMM	32-40-53	5	7/20/2006	CMM40-817
			SB	40-817-32-138	0	1/23/1985	40-817-32-138
			<i>To facilitate installation of transducers in aircraft axle.</i>				
40-821-2, 40-82197, 40-82113	WHEEL SPEED DETECTOR, HUBCAP ASSEMBLY, ADAPTER ASSEMBLY	??	CMM	32-47-05	0	9/15/2014	CMM40-821
			SB	40-817-32-61	0	3/27/1979	40-817-32-61
			<i>Improved method of transducer removal from aircraft.</i>				
40-835	40-835	B-727 MKIII	CMM	32-42-15	3	4/19/2008	CMM40-835
			SB	40-817-32-138	0	1/23/1985	40-817-32-138
			<i>To facilitate installation of transducers in aircraft axle.</i>				
40-837	HYTROL SKID DETECTOR ASSEMBLY	DHC-5, DHC-6	CMM	40-835-32-62	0	3/27/1979	40-835-32-62
			<i>Improved method of transducer removal from aircraft.</i>				
40-873, 40-937	HYTROL SKID DETECTOR AND HUBCAP ASSEMBLY	DHC-7	OHM	32-42-37	0	5/1/1976	OHM40-873
40-893	WHEEL SPEED TRANSDUCER	CESSNA 500	CMM	32-40-95	0	4/16/1982	CMM40-893
40-911	WHEEL SPEED TRANSDUCER (WST) ASSEMBLY	LEAR JET	CMM	32-40-77	2	4/19/2008	CMM40-911
40-955	TRANSDUCER ASSEMBLY	CESSNA 550,525	CMM	32-40-93	4	2/15/2016	CMM40-955

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40-979	WHEEL SPEED TRANSDUCER (WST) ASSEMBLY	Rockwell	CMM	32-43-93	0	7/31/2008	CMM40-979
			SB	40-979-6-8	0	11/1/1987	40-979-6-8
			<i>Application of locking compound to subject transducers.</i>				
40-999, -1	WHEEL SPEED TRANSDUCER	METROLINER	CMM	32-40-90	0	11/18/1980	CMM40-999
413800-01	FUEL SELECTOR VALVE	???	OHM	28-20-8	0	3/1/1968	413800-1
416090-1	FUEL SWING CHECK VALVE	???	OHM	28-20-10	0	3/1/1968	416090-1
42-033	HYTROL CONTROL SHIELD	B-707-320B	OHM	32-40-0	3	11/25/1968	OHM42-033
42-033-1	SKID CONTROL BOX ASSEMBLY	B-707-320B	OHM	32-40-0	0	6/1/1979	OHM42-033-1
			SB	8-127-1	2	8/5/1966	8-127-1
			<i>Eliminate possible HF interference.</i>				
			SB	8-134-1	0	1/15/1968	8-134-1
			<i>Incorporate a memory circuit permitting independent inboard and outboard locked wheel operation.</i>				
42-041	TBD	B-707/727	MAINT	32-40-0	2	8/26/1966	MAINT42-041
42-041-1	HYTROL MARK II SKID CONTROL BRAKING SYSTEM (SCBS) FIELD TEST SET	???	O&MM	N/A	1	5/1/2013	42-041-1
42-041-1 (EAR Version)	HYTROL MARK II SKID CONTROL BRAKING SYSTEM (SCBS) FIELD TEST SET	B-707/727	OMM	(No ATA Number)	1	5/1/2013	OMM42-041-1 (EAR Version)
42-041-1 (ITAR Version)	TBD	Various	OMM	(No ATA Number)	1	5/1/2013	OMM42-041-1 (ITAR Version)

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42-045,-1	HYTROL MARK II CONTROL BOX	BAC 1-11	CMM	32-40-98	4	5/1/2009	CMM42-045
			SB	8-108-1	0	5/15/1968	8-108-1
			<i>Product improvement.</i>				
42-051, 42-051P	UNIVERSAL AUTOBRAKE/ANTI-SKID TEST SET	VARIOUS	O&M	(No ATA Number)	4	6/3/1996	O&M42-051
			SB	42-051-32-37	1	3/1/1977	42-051-32-37
			<i>Eliminate circuit interaction during performance of control box functional test.</i>				
			SB	42-051P-32-28	3	5/29/1981	42-051P-32-28
			<i>Eliminate circuit interaction during performance of control box functional test.</i>				
			SB	8-121-3	0	6/20/1968	8-121-3
			<i>Provide correct active voltage when using cable assembly P/N 42-05128.</i>				
42-051100-1, 42-05120, 42-05123, 42-05126, 42-05128, 97-07310	TEST ADAPTER FOR THE BOEING 707-320/320B/320C/321C IRPLEANES ANTI SKID SYSTE	B-707	O&M	32-40-60	1	9/19/1980	O&M42-051100-1
42-051126, 42-051127, 42-051127-1, 42-051128, 42-051129	TBD	DC-9-20/30/40	O&M	32-40-58	1	6/15/1985	O&M42-051126
			SB	42-051126-32-74	0	5/12/1980	42-051126-32-74
			<i>Preclude resetting of test set lights when valve switches are actuated.</i>				
			SB	42-051127-32-58	0	2/15/1979	42-051127-32-58
			<i>Ensures XDCR loads during testing/calibration of 42-139335 main wheel PWA are same as in control box functional test.</i>				
			SB	42-051129-32-59	0	3/28/1979	42-051129-32-59
			<i>Ensures switch box wiring is equivalent to aircraft wiring configuration.</i>				
			SB	8-121-2	0	4/1/1968	8-121-2
			<i>Decouple AC ripple in regulated B+ voltage and improve simulated transducer loading effect.</i>				

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42-051157, 42-05117, 42-05126	HYTROL TEST ADAPTER ASSEMBLIES	DC-8-50/60	O&M	32-40-55	0	8/1/1971	O&M42-051157
42-05121, 42-05127, 42-05146, 42-051183, T-28400	HYTROL MARK II TEST ADAPTER ASSEMBLIES	B-727 MKII	O&M	32-40-59	1	12/16/1981	O&M42-05121
42-05127, 42-051408, 42-051467, 42-051662, 99-509, 99-659	HYTROL MARK III TEST SET ADAPTER ASSEMBLIES	B-727 MKIII	O&M	32-40-86	2	3/12/1984	O&M42-05127
42-051282, 42-051283, 42-051285, 42-051286, 99-035	TEST ADAPTER FOR THE DC-8 (63 SERIES) AIRPLANE ANTI SKID SYSTEM	B-727 MKII	O&M	32-40-56	1	9/26/1980	O&M42-051282

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42-051314, 42-051315, 42-051316, 42-051317, 42-051275	HYTROL TEST ADAPTER ASSEMBLIES	B-747	OMM	32-40-54	0	8/1/1971	OMM42-051314			
				SB	32-13	0	2/8/1972	32-13		
				<i>Modify cable switch box, P/N 42-051314-1 to test 42-189-7 skid control box.</i>						
				SB	32-14	1	5/18/1973	32-14		
				<i>Modify 42-051275 adapter cable 42-189-7 skid control box.</i>						
				SB	32-9	0	1/15/1971	32-9		
				<i>Inspect ad/or modify cable switch box, P/N 42-051314 to test 42-189-6 skid control box.</i>						
				SB	42-051275-2-32-128	0	12/1/1983	42-051275-2-32-128		
				<i>To prevent erroneous fault indications by simulating aircraft valve and wheel speed loads for intermittent connectors of P/N 42-051275-2 adapter cable.</i>						
				SB	42-051275-32-60	0	5/12/1979	42-051275-32-60		
<i>Adds capability to test primary and alternate tilt switch input circuits in 42-189-7 skid control box.</i>										
SB	42-051314-32-57	0	2/15/1979	42-051314-32-57						
<i>Adds capability to test qual locked wheel crossover circuits in 42-189-7 skid control box.</i>										
SB	42-051316-32-55	0	2/1/1979	42-051316-32-55						
<i>Adds capability to test/calibrate 42-189155 test card (Digital IC) used in 42-189-7 skid control box.</i>										
SB	42-051317-32-19	1	10/31/1974	42-051317-32-19						
<i>Adds capability to calibrate 42- 189132 main wheel cards used in 42-189-7 skid control box.</i>										
SB	42-051317-32-56	0	2/1/1979	42-051317-32-56						
<i>Adds capability to test/calibrate 42-189152 main wheel (Hybrid) cards used in 42-189-7 skid control box.</i>										
42-051408, 42-051410, 42-051412	HYTROL MARK III TEST CABLE SWITCH BOX ASSEMBLIES	B-737 MKIII	O&M	32-40-61	2	6/19/1980	O&M42-051408			
42-05145, 42-05148, 42-05149	HYTROL MARK II TEST ADAPTER ASSEMBLIES	DC-9-10	O&M	32-40-57	0	2/1/1972	O&M42-05145			
				SB	8-121-1	0	2/15/1967	8-121-1		
<i>Simulate impedance of dual pressure control valve at low temperature.</i>										

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42-051611, 42-051612, -1, 42-051668, -1	HYTROL MARK III TEST ADAPTERS AND CABLE SWITCH BOXES	DC-9-50/80, C9B	O&M	32-40-73	0	5/21/1980	O&M42-051611	
42-051636	TBD	LEARJET, JETSTAR-2	O&M	32-40-82	0	5/15/1977	O&M42-051636	
42-053, -A, -B, -1A, -1B	HYTROL MARK II BRAKE CONTROL UNIT (BCU)	DC-8-50/60	CMM	32-40-07	5	12/1/2022	CMM42-053	
			SB	32-1	0	4/9/1970	32-1	
			<i>Correct a condition caused by improper installation of capacitor C3 in spoiler amplifier printed wiring assembly P/N 42-053165.</i>					
			SB	32-5	1	9/1/1971	32-5	
			<i>Provide interchangeability of spoiler amplifier printed wiring assembly, P/N 42-05316, between control boxes.</i>					
			SB	42-053A-32-22	0	7/15/1974	42-053A-32-22	
			<i>Prevent incremental movement of ground spoiler actuator.</i>					
			SB	42-053A-32-47	0	12/15/1977	42-053A-32-47	
			<i>Eliminate possibility of a momentary brake release condition occurring upon antiskid turn on.</i>					
			SB	8-112-2 & 8-135-3	1	5/14/1969	8-112-2 & 8-135-3	
			<i>Rework lamp logic card P/N 42-051314 to preclude high frequency oscillation.</i>					
			SB	8-112-3 & 8-135-4	0	4/9/1970	8-112-3 & 8-135-4	
			<i>Rework spoiler amplifier card.</i>					
			SB	8-135-1	0	2/29/1968	8-135-1	
			<i>Eliminate possibility of inadvertent spoiler actuation.</i>					
42-073,-1	ANTI-SKID CONTROL UNIT	B-707-320C/321C	OHM	32-00-15	2	9/5/1986	OHM42-073	
			SB	8-114-1	1	8/5/1966	8-114-1	
			<i>Eliminate possible HF interference.</i>					
42-089-2	HYTROL SKID CONTROL BOX	DC-9-10	OHM	32-40-0	1	5/15/1966	OHM42-089-2	
			SB	8-103-1	2	2/23/1968	8-103-1	
			<i>Improve temperature stability.</i>					

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42-089-3	SKID CONTROL BOX	DC-9-10	OHM	32-40-0	3	2/23/1968	OHM42-089-3
			SB	8-138-1	1	3/31/1967	8-138-1
			<i>Eliminate failure indication on antiskid control panel.</i>				
42-089-4,-5,-6	SKID CONTROL BOX	DC-9-10	OHM	32-43-04	0	6/1/1979	OHM42-089-4
			SB	42-089-5-32-46	0	4/17/1978	42-089-5-32-46
			SB	8-149-1	0	4/10/1967	8-149-1
			<i>Delete time out function of control box.</i>				
			<i>Ensure proper internal voltage regulation with aircraft normal AC ripple and a signal failure induced transient condition on DC power input.</i>				
42-091 -1A	HYTROL SPOILER CONTROL BOX	DC-9, DC-9-80	OHM	27-60-01	0	8/31/1965	OHM42-091-1A
42-091-2	HYTROL MARK II SPOILER CONTROL UNIT (SCU)	DC-9	CMM	32-40-91	4	8/10/2011	CMM42-091-2
			SB	42-091-2-136-1	0	2/15/1968	42-091-2-136-1
			<i>Product improvement.</i>				
			SB	42-091-2-136-2	0	10/15/1968	42-091-2-136-2
			<i>Provide increased range for obtaining calibration voltages.</i>				
SB	42-091-2-32-90	0	8/28/1981	42-091-2-32-90			
			<i>Evaluation of C3 and C2 capacitors used on PWA's 42- 09144 and 42-09166 requires replacement of capacitors to meet design requirements.</i>				
SB	42-091-32-81	0	9/28/1980	42-091-32-81			
			<i>Determine proper clearance exists between diode terminal lugs.</i>				
42-091-3	HYTROL MARK II SPOILER CONTROL UNIT (SCU)	DC-9, MD-80	CMM	32-40-72	4	3/15/2008	CMM42-091-3
			SB	42-091-2-32-90	0	8/28/1981	42-091-2-32-90
			<i>Evaluation of C3 and C2 capacitors used on PWA's 42- 09144 and 42-09166 requires replacement of capacitors to meet design requirements.</i>				
SB	42-091-3-32-116	1	6/6/1987	42-091-3-32-116			
			<i>Modify hardware to meet DAC modified OATP (On Aircraft Test Procedure) and maintenance manual requirements for spoiler deployment.</i>				

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42-139-1,-2,-2A	SKID CONTROL BOX	DC-9-20/30/40	OHM	32-43-01	0	8/1/1970	OHM42-139-1		
			SB	32-10	0	5/1/1971	32-10		
			<i>Provide corrective action to eliminate possible false triggering of P/N 42-139-2A control box circuitry while testing aircraft.</i>						
			SB	32-11	0	8/25/1971	32-11		
			<i>Revise circuitry and change values of four capacitors to maintain required time constant.</i>						
			SB	32-15	0	4/14/1972	32-15		
			<i>Convenience to operator and DAC as compared to an alternative aircraft system change.</i>						
42-139-3,-4	HYTROL MARK III BRAKE CONTROL UNIT (BCU)	DC-9-20/30/40	CMM	32-43-07	4	3/16/2004	CMM42-139-3		
			SB	32-15	0	4/14/1972	32-15		
			<i>Convenience to operator and DAC as compared to an alternative aircraft system change.</i>						
			SB	42-139-32-70	0	5/15/1980	42-139-32-70		
			<i>Establish limits for new PBM decay requirements to prevent loss of braking during high speed taxi.</i>						
			SB	42-139-4-32-30	1	5/16/1977	42-139-4-32-30		
			<i>Adjust locked wheel crossover characteristic to preclude loss of brakes when both tires on one side of aircraft are blown.</i>						
SB	42-139-5-32-44	3	7/26/1979	42-139-5-32-44					
<i>Delete time out function of control box.</i>									

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42-139-5,-6	HYTROL MARK III CONTROL BOX	DC-9-20/30/40	CMM	32-43-09	4	5/30/2012	CMM42-139-5		
			SB	42-139-4_5_6_32-198	0	9/21/1990	42-139-4_5_6_32-198		
			<i>To update test and cutout card and calibration procedure for P/N 42-139-4/5/6 control units.</i>						
			SB	42-139-5-32-77	0	6/17/1980	42-139-5-32-77		
			<i>To eliminate possible oscillation of power supply.</i>						
42-147	SKID CONTROL BOX	CV-660	OHM	32-40-0	1	11/25/1968	OHM42-147		
			SL	42-147-8-141-1	0	4/24/1968	42-147-8-141-1		
			<i>(Description not available)</i>						
			CMM	32-40-10	2	9/19/1983	CMM42-157-1		
			OHM	32-40-0	1	6/30/1968	OHM42-159-1		
42-157-1	SKID CONTROL BOX	BAC 1-11	CMM	32-40-10	2	9/19/1983	CMM42-157-1		
42-159-1	SKID CONTROL BOX	JETSTAR-1	OHM	32-40-0	1	6/30/1968	OHM42-159-1		
42-189-6	HYTROL MARK III CONTROL BOX ASSEMBLY	B-747	OHM	32-42-01	1	5/25/1974	OHM42-189-6		
			SB	32-12	1	3/10/1972	32-12		
			<i>Improve performance of skid control system because of improved wheel speed resolution.</i>						
			SB	32-7	0	1/15/1971	32-7		
			<i>Modify auxiliary card printed assembly P/N 42-189104 used in 42-189-6 control box.</i>						
42-189-6	HYTROL MARK III CONTROL BOX ASSEMBLY	B-747	SB	32-8	0	1/15/1971	32-8		
			<i>Modify four test card printed wiring assemblies, P/N 42-189108 Rev A used in 42-189-6 control box.</i>						
			SB	42-189-6-32-9	0	1/15/1971	42-189-6-32-9		
<i>Describes the procedures to inspect and modify the 42-051314 switch box.</i>									

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42-189-7	HYTROL MARK III SKID CONTROL UNIT (SCU)	B-747	CMM	32-42-09	9	11/3/2009	CMM42-189-7
			SB	42-189-7-32-118	0	1/7/1983	42-189-7-32-118
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>				
			SB	42-189-7-32-51	0	10/1/1978	42-189-7-32-51
			<i>Cut B+1 network circuit to eliminate illumination of eight annunciator lamps in cockpit as landing gear is extended.</i>				
42-189-7	HYTROL MARK III SKID CONTROL UNIT (SCU)	B-747	SB	42-189-7-32-67	0	5/15/1980	42-189-7-32-67
			<i>Eliminate nuisance fault indications during low speed taxi and towing.</i>				
			SB	42-189-7-32-68	0	5/15/1980	42-189-7-32-68
<i>Introduce new production cards and eliminate fault indications during low speed taxi and towing.</i>							
42-197	TBD	SUMITOMO CXI	OHM	(No ATA Number)	0	9/15/1971	OHM42-197
42-201-1	TBD	LEAR 25	OHM	32-40-76	0	9/1/1975	OHM42-201-1
42-239	TBD	DHC-6	OHM	32-42-11	0	12/15/1972	OHM42-239
42-241	TBD	DC-8-63	CMM	32-40-03	0	7/10/2008	CMM42-241
42-243	SPOILER CONTROL UNIT	DC-8-63	OHM	27-60-04	3	5/2/1986	OHM42-243
			SB	42-243-27-1	0	8/25/1971	42-243-27-1
<i>Eliminate possibility of oscillations of transistors Q1 and Q2.</i>							
42-255	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-30/40, C9B	OHM	32-40-66	0	3/15/1974	OHM42-255
			SB	42-255-32-103	0	1/7/1983	42-255-32-103
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>				
42-255	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-30/40, C9B	SB	42-255-32-27	1	12/3/1982	42-255-32-27
			<i>Eliminate nuisance fault indication, adjust locked wheel crossover to preclude loss of brakes if both tires on one side of aircraft are blown.</i>				

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42-255-1	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-30/40, C9B	CMM	32-40-78	0	1/10/2012	CMM42-255-1		
			SB	42-255_443-32-69	1	11/18/1980	42-255_443-32-69		
			<i>To eliminate double testing of circuit when test switch is released.</i>						
			SB	42-255-1-32-102	0	1/7/1983	42-255-1-32-102		
<i>Provides improved reliability by reducing internal power consumption of regulator.</i>									
42-255-1	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-30/40, C9B	SB	42-255-1-32-41	2	12/3/1982	42-255-1-32-41		
			<i>Make control box assembly compatible with DACO antiskid/ spoiler test set P/N 5916819-503.</i>						
42-265	ANTISKID CONTROL UNIT (ACU)	727, 737	CMM	32-40-51	1	12/8/2020	CMM42-265		
			SB	42-265-32-120	0	1/7/1983	42-265-32-120		
			<i>Replacement of Voltage Regulator p/n 43-101 with p/n 43-101.</i>						
			SB	42-265-32-29	0	2/1/1976	42-265-32-29		
			<i>Modification and Reidentification of the 42-265 Control Box Assembly, by Metal Stamping the Suffix Letter "A" following the Unit Serial Number.</i>						
			SB	42-265-32-32	0	4/20/1976	42-265-32-32		
			<i>Describes the inspection/check and test necessary to determine if correct diodes are present at positions CR1, CR2, CR3 and CR4 in the A32 subassembly p/n 42-26598.</i>						
			SB	42-265-32-71	0	3/10/1980	42-265-32-71		
<i>(Description not available)</i>									
42-265	ANTISKID CONTROL UNIT (ACU)	727, 737	TR	TR32-40-51-02	0	1/25/2021	TR32-40-51-02		
			<i>Revises page 1051 and 1052, Figure 1020 (Sheet 1 of 2) and (Sheet 2 of 2).</i>						
			TR	TR32-40-51-03	0	1/25/2021	TR32-40-51-03		
			<i>Revises page 1085 and 1086, Figure 1034 (Sheet 1 of 2) and (Sheet 2 of 2).</i>						
42-265	ANTISKID CONTROL UNIT (ACU)	727, 737	TR	TR32-40-51-04	0	1/25/2021	TR32-40-51-04		
			<i>(Description not available)</i>						
42-269-1	TBD	DHC-6	OHM	27-60-05	0	9/1/1973	OHM42-269-1		



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
42-307	HYTROL MARK III CONTROL BOX	B-727 MKIII	OHM	32-42-10	2	1/28/1983	OHM42-307		
			SB	42-307-1-32-72	0	3/10/1980	42-307-1-32-72		
			<i>Replace wet tantalum capacitors on main wheel card to improve reliability.</i>						
			SB	42-307-32-121	0	1/7/1983	42-307-32-121		
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>						
42-307-1	AUTOBRAKE/ANTI-SKID CONTROL UNIT	B-727 MKIIIAB	SB	42-307-32-40	0	8/5/1976	42-307-32-40		
			<i>Eliminate possibility of velocity converter ripple building up current to antiskid control valve at low speeds.</i>						
			SL	42-307-1	0	9/5/1979	42-307-1		
			<i>(Description not available)</i>						
			CMM	32-42-31	0	8/1/2014	CMM42-307-1		
42-307-1	AUTOBRAKE/ANTI-SKID CONTROL UNIT	B-727 MKIIIAB	SB	42-307-1-32-104	0	1/7/1983	42-307-1-32-104		
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>						
			SB	42-307-1-32-39	0	7/2/1976	42-307-1-32-39		
			<i>Eliminate possibility of velocity converter ripple building up current to antiskid control valve at low speeds, causing a possible brake release.</i>						
			SB	42-307-1-32-72	0	3/10/1980	42-307-1-32-72		
<i>Replace wet tantalum capacitors on main wheel card to improve reliability.</i>									
42-307-1	AUTOBRAKE/ANTI-SKID CONTROL UNIT	B-727 MKIIIAB	SB	42-307-1-32-75	0	5/8/1981	42-307-1-32-75		
			<i>Provides improved comprehension monitoring of antiskid system and adds dynamic testing of antiskid control circuits within control unit.</i>						





PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
42-319-1, -3	AUTOBRAKE CONTROL UNIT	B-747	CMM	32-42-18	0	1/6/2009	CMM42-319-1		
			SB	42-319-1-32-124	0	3/17/1983	42-319-1-32-124		
								<i>Provides improved reliability by reducing internal power consumption of regulator.</i>	
			SB	42-319-1-32-21	0	3/15/1974	42-319-1-32-21		
								<i>Increase filtering time constant of antiskid fault monitoring input and increase maximum valve drive current for maximum deceleration setting.</i>	
			SB	42-319-1-32-25	0	11/19/1974	42-319-1-32-25		
								<i>Product improvement.</i>	
			SB	42-319-1-32-35	1	10/1/1977	42-319-1-32-35		
								<i>Re-tune control unit to reduce initial overshoot when MIN or MED is selected and prevent possible disarms due to a bounce after touchdown with MIN selected.</i>	
			SB	42-319-3-32-119	0	11/4/1982	42-319-3-32-119		
					<i>Provides improved reliability by reducing internal power consumption of regulator.</i>				
SB	42-319-3-32-139	1	6/20/1986	42-319-3-32-139					
					<i>To prevent possible latch-up after replacement of operational amplifier AR5.</i>				
SB	42-319-3-32-170	1	10/1/1991	42-319-3-32-170					
					<i>Product improvement. Improved stability of hybrid P/N 42-15570- 3 (Rev. K or later).</i>				
SB	42-319-3-32-95	1	11/16/1981	42-319-3-32-95					
					<i>Prevent autobrake disarm when antiskid system is off (landing gear up).</i>				
42-363	BRAKE CONTROL SYSTEM CONTROL BOX	METROLINER	CMM	32-40-88	0	11/18/1980	CMM42-363		
			SB	42-363-1-32-89	1	6/15/1984	42-363-1-32-89		
								<i>Provide a lighter weight housing and a conventional, circular connector to interface with aircraft wiring harness.</i>	
			SB	42-363-32-105	0	1/7/1983	42-363-32-105		
					<i>Provides improved reliability by reducing internal power consumption of regulator.</i>				
SB	42-363-32-88	1	6/24/1983	42-363-32-88					
					<i>Reduce heat buildup in control box and provide more protection against electromagnetic radio and frequency interference.</i>				

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42-375-1	SCBS HYTROL MARK III CONTROL BOX	JETSTAR-2	CMM	32-42-40	0	5/31/2013	CMM42-375-1		
			SB	42-375-1-32-140	0	2/4/1985	42-375-1-32-140		
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>						
42-379-1	TBD	LEAR 35	OHM	32-42-32	1	8/31/1998	OHM42-379-1		
42-397,-1, -2, -3	AUTOBRAKE CONTROL BOX	DC-10AB	CMM	32-44-20	0	12/1/2007	CMM42-397		
			SB	42-397-1-32-107	0	1/7/1983	42-397-1-32-107		
			SB	42-397-32-73	0	4/7/1980	42-397-32-73		
			<i>To preclude inadvertent master caution light illumination.</i>						
42-399	AUTOBRAKE CONTROL PANEL (ACP)	DC-10AB	CMM	32-44-02	2	7/15/2008	CMM42-399		
			SB	42-399-32-183	0	4/28/1989	42-399-32-183		
			<i>Provides more effective method of securing switch S2 stack.</i>						
42-409, -1, -2, -3	AUTOBRAKE CONTROL UNIT	DC-9AB	OHM	32-42-45	2	7/16/1990	OHM42-409		
			SB	42-409-1-32-159	1	9/15/1991	42-409-1-32-159		
								<i>Control units do not meet specification requirements for continued operation during power interrupts and transients.</i>	
			SB	42-409-1-32-91	0	8/28/1981	42-409-1-32-91		
								<i>To ensure closer control of individual decel rates and to provide decel rates compatible with DC-9-80.</i>	
			SB	42-409-2-32-190	0	4/12/1990	42-409-2-32-190		
								<i>Modifies P/N 42-409-2 into 42-409-3 to eliminate power transients and provide accurate adjustment of spoiler switches.</i>	
SB	42-409-32-108	0	1/7/1983	42-409-32-108					
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>						
SB	42-409-32-80	0	10/15/1980	42-409-32-80					
			<i>To preclude intermittent master caution light illumination when pilot takeover is initiated with brake pedal depression.</i>						

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42-411, 42-831	CONTROL PANEL AUTOBRAKE SELECT	DC-9AB, MD-80	OHM	32-42-44	3	9/16/1988	OHM42-411	
			SB	42-411-32-184	0	4/28/1989	42-411-32-184	
			<i>Provides more efficient method of securing switch S2 to mounting bracket.</i>					
42-443-1	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-50	OHM	32-42-29	1	3/24/1982	OHM42-443-1	
			SB	42-255_443-32-69	1	11/18/1980	42-255_443-32-69	
			<i>To eliminate double testing of circuit when test switch is released.</i>					
42-443-1	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-50	SB	42-443-1-32-109	0	1/7/1983	42-443-1-32-109	
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>					
			SB	42-443-1-32-42	0	5/4/1977	42-443-1-32-42	
42-443-1	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	DC-9-50	<i>Make control box assembly compatible with DACO antiskid/ spoiler test set P/N 5916819-503.</i>					
			SB	42-443-32-26	0	10/31/1975	42-443-32-26	
			<i>Eliminate nuisance fault indication of control box, adjust locked wheel crossover characteristics to preclude loss of brakes when both tires on one side of aircraft are blown and reduce stress on power supply components.</i>					
42-457-4	TBD	DHC-7	CMM	32-42-39	0	5/1/2012	CMM42-457-4	
42-471	HYTROL MARK III CONTROL BOX ASSEMBLY	DHC-5, BUFFALO	OHM	32-42-22	1	3/15/1976	OHM42-471	
42-511	TBD	BJET400	CMM	32-42-72	1	5/7/1986	CMM42-511	

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
42-527B, 42-527E, 42-527D, 42-527C	HYTROL MARK II CONTROL BOX	B-727 MKII	CMM	32-40-04	5	9/15/2005	CMM42-527		
			SB	42-527-32-43	0	4/1/1977	42-527-32-43		
			<i>Prevent possibility of brake release at antiskid turn on.</i>						
			SB	42-527-32-50	1	10/9/1979	42-527-32-50		
			<i>Improve temperature stability of locked wheel circuit.</i>						
			SB	42-527-32-64	0	5/25/1979	42-527-32-64		
			<i>Removes calibration requirements for R26 in valve driver circuit.</i>						
			SB	8-100-1	0	12/15/1964	8-100-1		
<i>Eliminate possibility of a false wheel indication of nose wheel.</i>									
42-547-2	ANTI-SKID CONTROL UNIT	CESSNA 550	CMM	32-42-63	0	4/26/1989	CMM42-547-2		
			SB	42-547-2-32-115	1	8/17/1983	42-547-2-32-115		
			<i>To reduce possibility of brake pressure reduction with elevated temperatures due to component characteristics within control box.</i>						
42-607-1	HYTROL MARK IIIA SKID CONTROL UNIT (SCU)	MD-80, DC 9-80	CMM	32-42-49	7	10/14/2011	CMM42-607-1		
			SB	42-607-1-32-111	0	1/7/1983	42-607-1-32-111		
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>						
			SB	42-607-1-32-145	5	2/1/2004	42-607-1-32-145		
			<i>Douglas MD-80 series aircraft equipped with or retrofitted to Bendix brake require use of H-A P/N 42-807 antiskid control unit.</i>						
SB	42-607-1-32-155	0	3/11/1987	42-607-1-32-155					
<i>1. To eliminate "double" test cycle. 2. To ensure that a complete test cycle occurs during power applications.</i>									
SB	42-607-1-32-4	2	7/14/1997	42-607-1-32-4					
<i>Modification of Wheels Not Rolling Card 42-60719 to 42-607191</i>									

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.							
42-625-1	TBD	AERMACCHI MB339	CMM	32-40-44	0	6/17/1988	CMM42-625-1							
42-629,-1,-2	HYTROL MARK IV ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	B-767	CMM	32-42-52	9	1/30/2012	CMM42-629							
								ATLAS	N/A	R	1/8/2001	ATL757		
								SB	42-629-1-32-153	0	4/10/1987	42-629-1-32-153		
								<i>Followup action to Boeing Operators Manual Bulletin 767-86-8.</i>						
								SB	42-629-1-32-180	0	2/24/1989	42-629-1-32-180		
								<i>Provides instructions to modify control unit P/N 42-629-1 to 42-877-1.</i>						
								SB	42-629-2-32-181	0	2/28/1989	42-629-2-32-181		
<i>Provides instructions to modify control unit P/N 42-629-2 to 42-877-1.</i>														
42-639,-1,-2	TBD	DC-9-80AB	CMM	32-42-50	1	1/10/1990	CMM42-639							
								SB	42-639-1-32-160	1	9/15/1991	42-639-1-32-160		
								<i>Control units do not meet specification requirements for continued operation during power interrupts and transients.</i>						
								SB	42-639-1-32-92	0	8/28/1981	42-639-1-32-92		
								<i>To ensure closer control of individual decel rates and to provide decel rates compatible with DC-9-80.</i>						
								SB	42-639-2-32-191	0	4/12/1990	42-639-2-32-191		
								<i>Modifies P/N 42-639-2 to 42-639-3 to eliminate power transients and provide accurate adjustment of spoiler switches.</i>						
SB	42-639-32-112	0	1/7/1983	42-639-32-112										
<i>Provides improved reliability by reducing internal power consumption of regulator.</i>														
42-639,-1,-2	TBD	DC-9-80AB	CMM	32-42-50	1	1/10/1990	CMM42-639							
								SB	42-639-32-83	0	10/15/1980	42-639-32-83		
<i>To preclude intermittent master caution light illumination when pilot takeover is initiated with brake pedal depression.</i>														

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
42-641, 42-641-1, 42-641-2	ANTISKID AUTOBRAKE CONTROL UNIT (AACU)	B-757	CMM	32-42-59	8	10/4/2019	CMM42-641		
			ATLAS	N/A	R	1/8/2001	ATL757		
			SB	42-641-1-32-152	2	8/10/1988	42-641-1-32-152		
			<i>Followup action to Boeing Operators Manual Bulletin 757-85-2.</i>						
			SB	42-641-2-32-2	0	10/30/2007	42-641-2-32-2		
			<i>To allow existing AACUs P/N 42-641-2 to be modified for use on 757 aircraft equipped with carbon brakes; P/N 42-641-2 to 42-757-1.</i>						
42-651-01	ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	B-737-200	CMM	32-40-87	7	10/2/2017	CMM42-651-01		
			SB	42-651-01-32-122	0	1/7/1983	42-651-01-32-122		
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>						
			SB	42-651-32-94	0	11/2/1981	42-651-32-94		
<i>Incorporates several program changes which will provide improved autobrake operational reliability.</i>									
42-663	HYTROL MARK III AUTOBRAKE CONTROL UNIT (ACU)	B-727	CMM	32-42-70	1	3/1/2004	CMM42-663		
			SB	42-663-32-173	0	6/27/1988	42-663-32-173		
<i>Replaces R13 resistor network on BITE interface card. Reduces nominal values of R13A - R13H from 10 to 1 kilohms.</i>									
42-665	AUTOBRAKE CONTROL PANEL (ACP)	DC-10AB (JAL)	CMM	32-44-03	1	10/15/2008	CMM42-665		
42-669-1	HYTROL MARK III SKID CONTROL UNIT (SCU)	CESSNA 650	CMM	32-42-73	4	11/5/2006	CMM42-669-1		

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42-683-2, -3	HYTROL MARK III SCBS CONTROL BOX ASSEMBLY	SAAB 340	CMM	32-43-20	4	7/20/2008	CMM42-683-2			
			SB	42-683-2-32-1	0	8/16/1995	42-683-2-32-1			
<i>To provide wheel spin-up discrete outputs for other aircraft systems.</i>										
42-687-1	HYTROL MARK III SKID CONTROL UNIT (SCU)	EMB-120	CMM	32-42-75	2	1/31/2008	CMM42-687-1			
42-695, -1	HYTROL MARK III CONTROL BOX ASSEMBLY	Beech 1900	CMM	32-46-02	0	11/25/2009	CMM42-695			
42-707-1, -2	HYTROL MARK III SCBS CONTROL BOX	CESSNA 550ADV, 525	CMM	32-42-95	4	1/1/2016	CMM42-707-1			
			SB	42-707-1-32-151	0	5/16/1985	42-707-1-32-151			
<i>Operators may have experienced antiskid control unit pressure bias modulation (PBM) build-up during low speed/taxi conditions. This could have resulted in brake pressure reduction at low speeds.</i>										
42-713	TBD	B-737-200	CMM	32-42-81	1	12/31/2004	CMM42-713			
42-719-01	HYTROL MARK III SKID CONTROL BOX	B-737-300	CMM	32-42-90	3	9/1/2014	CMM42-719-01			
			SB	42-719-01-32-147	2	6/25/1990	42-719-01-32-147			
			<i>To eliminate momentary brake release caused by power application to control unit.</i>							
			SB	42-719-01-32-157	0	4/13/1987	42-719-01-32-157			
<i>To eliminate momentary brake release when power is applied to control unit.</i>										
42-719-02	HYTROL MARK III SCBS CONTROL BOX	B-737-300/-400/-500	CMM	32-42-93	6	9/1/2014	CMM42-719-02			
			SB	42-719-01-32-187	0	7/5/1989	42-719-01-32-187			
			<i>Provides instructions to modify control unit P/N 42-719-01 into 42-719-02.</i>							

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42-741	HYTROL MARK III SKID CONTROL UNIT (SCU)	ATR-42	CMM	32-42-86	6	8/15/2016	CMM42-741			
			SB	42-741-32-175	0	11/11/1988	42-741-32-175			
<i>Modifies lower cover assembly P/N 42-74131 by replacing existing soft-sponge pads with composite-sponge pads.</i>										
42-747, -1, -2, -3	BRAKE SYSTEM CCONTROL UNIT (BCU)	B-747-400	CMM	32-42-94	8	9/16/2021	CMM42-747			
			ATLAS	N/A	C	12/23/1991	ATL 747			
			ATLAS	N/A	J	9/24/1993	ATL 747-2			
			ATLAS	N/A	E	9/24/1993	ATL 747-3			
			SB	42-747-2-32-196	0	1/21/1992	42-747-2-32-196			
<i>To upgrade BITE capabilities and incorporate minor hardware updates to improve EMI characteristics.</i>										
42-757	TBD	SABRELINER 65	OP HBK	(No ATA Number)	0	2/25/1980	OP-HBK42-757			
42-757,-1	HYTROL MARK IV ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	B-757	CMM	32-42-91	6	10/10/2006	CMM42-757			
			ATLAS	N/A	R	1/8/2001	ATL757			
			SB	42-757-1-32-205	4	6/24/2005	42-757-1-32-205			
			<i>To allow existing control units, P/N 42-757-1, to be modified (to 42-641-2) for use on 757 aircraft equipped with steel brakes.</i>							
			SB	42-757-32-154	0	4/10/1987	42-757-32-154			
<i>Describes the procedure to modify and reidentify P/N 42-75723 autobrake cards.</i>										



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42-767, 42-767-1, 42-767-2; 42-877, 42-877-1	ANTISKID/AUTOBRAKE CONTROL UNIT (ACU)	B-767	CMM	32-42-20	18	11/4/2022	CMM42-767_42-877		
			ATLAS	N/A	R	1/8/2001	ATL757		
			SB	42-767-1-32-1	2	6/30/2022	42-767-1-32-1		
			<i>Conversion of P/N 42-767, 42-767-1 or 42-877-1 Antiskid/ Autobrake Control Unit to P/N 42-767-2</i>						
			SB	42-767-32-182	0	2/24/1989	42-767-32-182		
			<i>Provides instructions to modify control unit P/N 42-767 into 42-877-1</i>						
			SB	42-767-32-208	0	5/30/1991	42-767-32-208		
			<i>Conversion to control unit P/N 42-767-1 from control units 42-767 or converted 42-877-1 to 42-767 control units for B-767 aircraft equipped with steel brakes.</i>						
42-791-1A, 42-867, 42-881	HYTROL MARK III SKID CONTROL UNIT	DHC-8, -300, -300A	CMM	32-42-83	4	10/5/2006	CMM42-791-1A		
			SB	42-791-1-32-164	0	6/1/1987	42-791-1-32-164		
			<i>Prevent possible flashing of antiskid annunciation at gear extension.</i>						
42-801, 42-801-1	HYTROL MARK III SKID CONTROL UNIT (SCU)	ATR-42, ATR 72	CMM	32-42-25	3	12/15/2016	CMM42-801		
42-807	HYTROL MARK IIIA CONTROL BOX ASSEMBLY	MD-80	CMM	32-42-04	4	7/30/2008	CMM42-807		
			SB	42-807-32-1	0	3/17/1997	42-807-32-1		
			<i>Modification of Wheels Not Rolling Card 42-80722 to 42-606191</i>						
			SB	42-807-32-163	0	5/8/1987	42-807-32-163		
<i>To ensure that a test cycle occurs during power application.</i>									
			SB	42-807-32-178	1	2/16/1989	42-807-32-178		
<i>To eliminate double test cycle, by adding R108 and R109 to test card 42-80719.</i>									

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
42-809,-2,-3	AUTOBRAKE CONTROL BOX	MD-80	CMM	32-42-56	5	11/2/2011	CMM42-809
			SB	42-809-2-32-189	0	4/12/1990	42-809-2-32-189
			<i>Modifies P/N 42-809-2 to 42-809-3 to eliminate power transients, provide accurate adjustment of spoiler switches and correct flap settings on BITE placard.</i>				
			SB	42-809-32-113	0	1/7/1983	42-809-32-113
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>				
42-839,-1,-2	AUTOBRAKE CONTROL BOX	DC-9-80, MD-80	CMM	32-42-57	4	3/16/2012	CMM42-839
			SB	42-839-1-32-192	0	4/12/1990	42-839-1-32-192
			<i>Modified 42-839-1 to 42-839-2 to eliminate power transients, provide accurate adjustment of spoiler switches and correct flap settings on BITE placard.</i>				
			SB	42-839-32-114	0	1/7/1983	42-839-32-114
			<i>Provides improved reliability by reducing internal power consumption of regulator.</i>				
42-853, 42-853-1	HYTROL SELECT DECEL AUTOBRAKE CONTROL UNIT (ACU)	MD-11	CMM	32-42-21	6	4/23/2019	CMM42-853
			SB	42-853-32-212	0	4/1/1992	42-853-32-212
			<i>Modify basic to -1.</i>				
42-885-2,-3	HYTROL MARK III SKID CONTROL UNIT (SCU)	CRJ-100/200	CMM	32-43-23	4	12/15/2016	CMM42-885-2
			SB	42-885-2-32-1	0	4/27/1994	42-885-2-32-1
<i>Modifies -2 control unit to -3 configuration.</i>							
42-889, 42-889-1	ANTISKID CONTROL UNIT ASSEMBLY	BAE JETSTREAM 41	CMM	32-43-25	1	1/15/1996	CMM42-889-1

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42-897-1	TBD	SAAB 2000	CMM	32-43-22	0	6/30/1994	CMM42-897-1			
42-901	TBD	BJET400A	CMM	32-43-24	1	9/20/2008	CMM42-901			
42-907-1, -2, -3, -4	BRAKE SYSTEM CONTROL UNIT (BSCU)	B-777	CMM	32-43-41	8	5/31/2019	CMM42-907			
			ATLAS	N/A	H	8/4/2004	AT42907			
			SB	42-907-1-32-1	0	11/4/1997	42-907-1-32-1			
			<i>Modifies P/N 42-907-1 to 42-907-2</i>							
			SB	42-907-2-32-1	0	7/3/2002	42-907-2-32-1			
			<i>Provides instructions to upgrade the 42-907-1 or 42-907-2 BSCU to the 42-907-3 configuration.</i>							
42-913, 42-913-1	HYTROL MARK III SCBS CONTROL BOX	CESSNA 750	CMM	32-43-54	1	6/11/2020	CMM42-913-1			
			42-925-1	HYTROL MARK III SKID CONTROL UNIT (SCU)	CANADAIR CL604	CMM	32-43-53	5	6/9/2020	CMM42-925-1
			SB			42-925-1-32-1	0	6/9/2020	42-925-1-32-1	
			<i>Conversion of Skid Control Unit, p/n 42-925-1 by modification of the bite control PCB assembly.</i>							
TR	TR32-43-53-01	0	10/12/2020			TR32-43-53-01				
<i>Revises Table 1004.</i>										
42-927, -1, -2, -3	DIGITAL MARK IV SCBS CONTROL BOX	CESSNA 560	CMM	32-43-57	4	7/15/2021	CMM42-927-1			
			SB	42-927-32-1	0	5/6/1996	42-927-32-1			
			<i>Modifies 42-927 to 42-927-1</i>							

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42-92750	ANTISKID FAULT DISPLAY UNIT	CESSNA 560	CMM	32-43-58	0	7/31/1997	CMM42-92750
			TR	TR32-43-58-01	0	11/21/2016	TR32-43-58-01
			<i>(Description not available)</i>				
42-933-1,-2,-3	HYTROL MARK V BRAKE CONTROL UNIT (BCU)	LEAR 45	CMM	32-43-47	4	6/10/2021	CMM42-933-1
			SB	42-933-2-32-1	0	5/18/2000	42-933-2-32-1
			<i>Eliminate potential for uncommanded braking</i>				
42-935, 42-935-2	ANTISKID/AUTOBRAKE CONTROL UNIT (AACU)	B-737-700/737-800	CMM	32-43-65	7	3/30/2023	CMM42-935
			ATLAS	N/A	8	4/17/2002	AT42935
			SB	42-935-2-32-1	0	8/4/2012	42-935-2-32-1
			<i>Provides instructions to return the AACUs to Crane Aerospace &amp; Electronics, Hydro-Aire, Inc. for transistor replacement on attrition basis only.</i>				
			SB	42-935-32-1	0	6/28/1999	42-935-32-1
			<i>To correct overratteting Selector Switch</i>				
			SB	42-935-32-2	5	10/23/2015	42-935-32-2
			<i>P/N 42-935-2 AACU's are tuned for steel brakes and cannot be used with carbon brakes. The SB provides instructions to convert 42-935-2 AACU'S to 142-147 for use on the carbon brake equipped 737NG aircraft.</i>				

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
42-951-1,-2,-3,-4,-5,-6	HYTROL MARK V BRAKE CONTROL UNIT (BCU)	EMB-145, 145-AEW&C	CMM	32-43-50	6	11/19/2010	CMM42-951-1		
			SB	42-951-1-32-1	0	3/31/1999	42-951-1-32-1		
			<i>Modifies 42-951-1 to -2</i>						
			SB	42-951-1-32-3	0	6/4/2004	42-951-1-32-3		
			<i>Modification to convert P/N 42-951-3 BCU to 42-951-5 BCU.</i>						
			SB	42-951-2-32-1	0	4/21/2000	42-951-2-32-1		
			<i>Corrects an uncommanded braking potential. This updates the P/N 42-951-2 BCU to the P/N 42-951-3 configuration.</i>						
			SB	42-951-5-32-1	1	5/4/2010	42-951-5-32-1		
			<i>Material part number correction.</i>						
			TR	TR32-43-50-01	0	1/27/2015	TR32-43-50-01		
<i>(Description not available)</i>									
			TR	TR32-43-50-02	0	1/27/2015	TR32-43-50-02		
<i>(Description not available)</i>									
			TR	TR32-43-50-03	0	1/27/2015	TR32-43-50-03		
<i>(Description not available)</i>									
			TR	TR32-43-50-04	0	1/27/2015	TR32-43-50-04		
<i>(Description not available)</i>									
42-955, -1	HYTROL MARK IV ANTISKID CONTROL UNIT (ACU)	GULFSTREAM V	CMM	32-45-01	4	3/20/2016	CMM42-955		
			SB	42-955-32-1	0	8/12/2005	42-955-32-1		
<i>On aircraft testing of BCU P/N 42-955 to ensure the BCU has the correct capacitors installed in locations C1, C2, C3, and C4 on the motherboard P/N 42-95533.</i>									



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
42-965,-1,-2, -3, -4	HYTROL MARK V DIGITAL BRAKE-BY-WIRE BRAKE CONTROL UNIT (BCU)	BD700 (GX)	CMM	32-43-69	8	10/26/2022	CMM42-965		
			SB	42-965-1-32-1	1	2/10/2005	42-965-1-32-1		
			<i>Modification to convert Brake Control Unit from P/N 42-965-1 to -2.</i>						
			SB	42-965-2-32-1	0	11/1/2012	42-965-2-32-1		
			<i>To provide instructions for transistor replacement for BCUs that are returned to Crane Aerospace &amp; Electronics on attrition only.</i>						
			SB	42-965-2-32-2	1	6/20/2012	42-965-2-32-2		
			<i>To provide instructions to return the BCUs to Crane Aerospace for R37 replacement.</i>						
			SB	42-965-2-32-3	0	10/8/2013	42-965-2-32-3		
<i>To provide rework instructions to upgrade P/N 42-965-2 to P/N 42-965-3.</i>									
SB	42-965-2-32-4	0	4/9/2013	42-965-2-32-4					
<i>To replace incorrect zener diodes VR5 through VR20, VR24 through VR31, and VR42 on Wheel Control Printed Circuit Board (PCB) assemblies A1 and A2.</i>									
SB	42-965-2-32-5	0	8/28/2013	42-965-2-32-5					
<i>To provide instructions to return the BCUs to Crane Aerospace for inspection and replacement of faulty transistors as required.</i>									
SB	42-965-3-32-1	0	11/3/2021	42-965-3-32-1					
<i>Upgrades BCU PN 42-965-3 to PN 42-965-4 by reworking the PCB assemblies to incorporate hardware changes on wheel control PCB assemblies A1 and A2.</i>									
42-977	BRAKE CONTROL UNIT (BCU)	B-717	CMM	32-45-01	5	9/7/2021	CMM42-977		
			SB	SB42-977-32-1	0	4/15/2000	SB42-977-32-1		
			<i>Rework of Brake Control Unit (BCU) wheel control and Autobrake circuit cards.</i>						
SB	SB42-977-32-2	0	5/28/2003	SB42-977-32-2					
<i>To prevent PCB assemblies XA1 and XA2 from sliding out.</i>									

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42-985, -1, -2	DIGITAL MARK IV ANTISKID CONTROL UNIT (ACU)	CESSNA 560 XL	CMM	32-41-07	1	4/14/2005	CMM42-985		
			TR	TR_32-41-07-01	0	8/18/2015	TR_32-41-07-01		
			<i>(Description not available)</i>						
			TR	TR32-41-07-02	0	8/18/2015	TR32-41-07-02		
			<i>Revises page 10021, IPL Figure 3</i>						
			TR	TR32-41-07-03	0	10/21/2016	TR32-41-07-03		
			<i>Revises page 10019, IPL Figure 3. Update Item 15 and add Item -15A.</i>						
42-989, -1	DIGITAL MARK IV CONTROL BOX	PREMIER I	CMM	32-43-74	2	4/5/2013	CMM42-989		
43-201, 43-001, 43-101	VOLTAGE REGULATOR	????	SL	43-201-6-82	0	12/9/1982	43-201-6-82		
			<i>(Description not available)</i>						
48-045	TBD	B-707	OHM	29-20-64	0	3/5/1980	OHM48-045		
48-077	TBD	ATR-42, ATR-72	CMM	32-42-89	0	3/16/1984	CMM48-077		
52-551	TBD	AERMACCHI MB339	CMM	32-43-34	0	5/15/1992	CMM52-551		
52-567	TBD	GULFSTREAM V	CMM	32-45-04	1	4/20/2008	CMM52-567		
52-61514314	PRESSURE SWITCH	ERJ170, ERJ190	CMM	32-44-75	0	10/7/2022	CMM52-61514314		
52-615500, 52-615500-1	PARKING BRAKE PRESSURE SWITCH	ERJ-170/-190	CMM	32-44-09	4	5/31/2019	CMM52-615500		
52-621500, -1	AUTOBRAKE SWITCH	ERJ-170/-190	CMM	32-41-19	3	1/18/2019	CMM52-621500		
52-633500, 52-633500-1	PRESSURE TRANSDUCER	ERJ-170/-190	CMM	32-41-05	3	10/1/2016	CMM52-633500		
52-659500	TBD	C919	ACMM	32-41-07	0	8/30/2019	ACMM52-659500		
52-665500	BRAKE TEMPERATURE PROBE (BTP)	C919	ACMM	32-44-02	0	8/20/2019	ACMM52-665500		

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52-669	TBD	C919	ACMM	32-41-11	0	8/30/2019	ACMM52-669
52-67914314	PRESSURE SWITCH	ERJ170, ERJ190	CMM	32-44-76	0	4/21/2023	CMM52-67914314 R
52-679500-1	PRESSURE SWITCH	ERJ170, ERJ190	ACMM	32-44-77	0	6/27/2023	ACMM_52-679500-1 R
52-685500	INTEGRATED PEDAL TRANSDUCER (IPT)	C919	ACMM	32-41-02	0	9/13/2019	ACMM52-685500
52737	LANDING GEAR SEQUENCE VALVE	B-707	OHM	29-20-60	0	12/20/1979	OHM52737
53047	TBD	HAMILTON 63E60 PROPELLER	OHM	29-20-67	0	1/15/1974	OHM53047
53814	HYDRAULIC SHUTTLE VALVE	B-707	OHM	29-20-61	0	5/1/1975	OHM53814
55962-45	SOLENOID 4-WAY HYDRAULIC SELECTOR VALVE WITH MANUAL OVERRIDE	??	SB	55962-45	0	9/1/1976	55962-45
							<i>(Description not available)</i>
60-611, -A	FUEL BOOSTER PUMP	DHC	OHM	28-20-0	1	2/15/1971	OHM60-611
60-649	FUEL PUMP ASSEMBLY	DC-8	OHM	28-10-01	1	4/1/1968	OHM60-649
60-651B	FUEL BOOSTER PUMP	DHC-5	OHM	28-20-27	2	5/30/1981	OHM60-651B

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60-70303, -3, 60-70306, -1, -3, 60-70307, -3, 60-70308, -3, 60-70309, -3, 60-703103, -3, 60-703104, -3, 60-703113, -1, -3, 60-72301, -3, -4, -5	FUEL JETTISON PUMP MOTOR-IMPELLER SUBASSEMBLY	B-747	CMM	28-31-05	2	7/26/2007	CMM60-703XX		
			SB	60-70300-28-1	0	12/21/2005	60-70300-28-1		
			<i>To convert the motor-impeller subassemblies 60-703XX and 60-723XX Series P/N's to the 60-72301-5 configuration.</i>						
			SB	60-703-2-28-4	0	1/15/1973	60-703-2-28-4		
			<i>Evacuate entrapped air from motor cavity to increase serviceability of pump.</i>						
			SB	60-703-2-28-7	0	1/15/1973	60-703-2-28-7		
			<i>(Description not available)</i>						
60-703200, 60-703200-1, 60-703200-2	FUEL JETTISON PUMP HOUSING SUBASSEMBLY	B-747, 747-400, 747-8	CMM	28-31-04	3	1/11/2018	CMM60-703200		
			SB	60-703_723-28-2	0	12/21/2005	60-703_723-28-2		
			<i>To replace the special washer used in the inlet check valve flapper mounting hardware with a new special washer that has a different contour. The modification results in decreased wear of the inlet adapter and therefore longer service life.</i>						
			SL	60-703200--TR1	0	4/20/2001	60-703200--TR1		
			<i>(Description not available)</i>						

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-703A, 60-703-2, -3, -4	FUEL JETTISON/OVERRIDE PUMP	B-747	OHM	28-31-01	4	7/15/1992	OHM60-703
			SB	60-703_723-28-2	0	12/21/2005	60-703_723-28-2
			<i>To replace the special washer used in the inlet check valve flapper mounting hardware with a new special washer that has a different contour. The modification results in decreased wear of the inlet adapter and therefore longer service life.</i>				
			SB	60-703-1-28-1	0	5/1/1970	60-703-1-28-1
			<i>Modification of Flapper Assembly.</i>				
			SB	60-703-28-24	0	1/6/1988	60-703-28-24
			<i>Improved inlet check valve assembly.</i>				
			SB	60-703-28-32	1	2/15/1999	60-703-28-32
			<i>Replacement of Thrust Washer P/N 60-06561</i>				
			SB	60-703-28-33	2	2/17/2003	60-703-28-33
			<i>Replaces inlet check valve to prevent excessive wear. Rev 1 adds pricing.</i>				
			SB	60-703-28-34	1	11/20/2000	60-703-28-34
			<i>Replaces discharge check valve to increase its structural integrity.</i>				
			SB	60-703-28-36	0	4/16/2002	60-703-28-36
			<i>Documents modifications required to convert 60-703 Series Fuel Pump Motor-Impeller Subassemblies to the improved 60-72301-4 config., which incorporates a new end cap &amp; pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminant</i>				
			SB	60-703-3-28-10	0	9/15/1974	60-703-3-28-10
			<i>Improve capability of thermal fuses to sense stator winding temperature and improve thermal fuse reaction time.</i>				
			SB	60-703-3-28-20	0	7/30/1980	60-703-3-28-20
			<i>Improve pump removal inlet check valve.</i>				
			SB	60-703-3-28-7	1	11/30/1973	60-703-3-28-7
			<i>Increase cooling to motor section and improve performance for operation under adverse conditions.</i>				
			SB	60-703-3-28-8	0	2/27/1973	60-703-3-28-8
			<i>Install handle assembly to protect cable and provide improved electrical connection to stator.</i>				
			SB	60-703-5-28-11	3	9/7/1979	60-703-5-28-11
			<i>Extend pump reliability and outline changes to provide improved altitude performance for pumping units installed in center wing tank using JP-4 fuel.</i>				



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-703A, 60-703-2, -3, -4	FUEL JETTISON/OVERRIDE PUMP	B-747	SB	60-703-5-28-12	2	3/25/1976	60-703-5-28-12
			<i>Extend pump reliability.</i>				
			SL	60-703-8-19	0	8/19/1988	60-703-8-19
			<i>(Description not available)</i>				
			TR	TR28-31-01-01	0	7/31/1997	TR28-31-01-01
			<i>(Description not available)</i>				
			TR	TR28-31-01-02	0	7/31/1997	TR28-31-01-02
			<i>(Description not available)</i>				
			TR	TR28-31-01-03	0	1/8/1999	TR28-31-01-03
			<i>(Description not available)</i>				
			TR	TR28-31-01-04	0	7/15/1999	TR28-31-01-04
			<i>(Description not available)</i>				
			TR	TR28-31-01-5	0	7/15/1999	TR28-31-01-5
			<i>(Description not available)</i>				



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-721, 60-721A, 60-723, 60-723A, 60-72101, 60-72101-1, 60-72101-3, 60-72101-3A, 60-72301, 60-70305, 60-70305-1, 60-703200	FUEL OVERRIDE/JETTISON PUMP ASSEMBLY, MOTOR-IMPELLER SUBASSEMBLY AND HOUSING SUBASSEMBLY	B-747	CMM	28-31-02	11	1/31/2004	CMM60-721_60-723
			SB	60-703_723-28-2	0	12/21/2005	60-703_723-28-2
			<i>To replace the special washer used in the inlet check valve flapper mounting hardware with a new special washer that has a different contour. The modification results in decreased wear of the inlet adapter and therefore longer service life.</i>				
			SB	60-703-28-33	2	2/17/2003	60-703-28-33
			<i>Replaces inlet check valve to prevent excessive wear. Rev 1 adds pricing.</i>				
			SB	60-703-28-34	1	11/20/2000	60-703-28-34
			<i>Replaces discharge check valve to increase its structural integrity.</i>				
			SB	60-703-28-35	3	2/3/2004	60-703-28-35
			<i>Replaces inlet adapter to prevent excessive wear. Rev 1 adds pricing. Rev 3 updates pricing info for kits.</i>				
			SB	60-703-28-36	0	4/16/2002	60-703-28-36
			<i>Documents modifications required to convert 60-703 Series Fuel Pump Motor-Impeller Subassemblies to the improved 60-72301-4 config., which incorporates a new end cap &amp; pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminant</i>				
			SB	60-72101-28-1	3	3/9/2006	60-72101-28-1
			<i>Documents modifications required to convert 60-72101 Series Fuel Pump Motor-Impeller Subassemblies to the improved 60-72101-4 config., which incorporates a pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-721-28-23	0	9/30/1982	60-721-28-23
			<i>Establishes configuration control for converted pumps due to duplication of serial numbers.</i>				
			SB	60-721-28-3	1	2/15/1999	60-721-28-3
			<i>Replacement of Thrust Washer P/N 60-06561</i>				
			SB	60-72301-28-1	0	4/16/2002	60-72301-28-1
			<i>Documents modifications required to convert 60-72301/60-72301-3 Series Fuel Pump Motor-Impeller Subassemblies to the improved 60-72301-4 config. Incorporates a new end cap &amp; pottingless connector. Prevents accum. of corrosion/conductive fuel contaminants.</i>				
			SB	60-723-28-5	3	2/3/2004	60-723-28-5
			<i>Replaces inlet adapter to prevent excessive wear. Rev 1 adds pricing. Rev 3 updates pricing info for kits.</i>				

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-721, 60-721A, 60-723, 60-723A, 60-72101, 60-72101-1, 60-72101-3, 60-72101-3A, 60-72301, 60-70305, 60-70305-1, 60-703200	FUEL OVERRIDE/JETTISON PUMP ASSEMBLY, MOTOR-IMPELLER SUBASSEMBLY AND HOUSING SUBASSEMBLY	B-747	SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01
				<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>			
			SB	SB60-721-28-5	3	2/3/2004	SB60-721-28-5
				<i>Replaces inlet adapter to prevent excessive wear. Rev 1 adds pricing. Rev 3 updates pricing info for kits.</i>			
			SL	60-703-2-12	0	3/1/1999	60-703-2-12
				<i>(Description not available)</i>			
			SL	SIL28-22-Aeroshell	0	12/10/2007	SIL28-22-Aeroshell
				<i>Announcement of BMS 3-33 (Aeroshell Grease 33) as substitute for Aeroshell Grease 16</i>			
			SL	SIL28-22-Epocast	0	4/25/2012	SIL28-22-Epocast
				<i>Announce discontinuance of support for legacy motor-impeller subassemblies and cables which use Epocast 220A potting compound.</i>			

R (RIGHT MARGIN) -- INDICATES A DOCUMENT THAT HAS BEEN UPDATED OR ADDED TO THIS INDEX IN THE PAST 12 MONTHS.



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-72101-2, -3B, -4, -5	FUEL JETTISON PUMP MOTOR-IMPELLER SUBASSEMBLY	B-747, 747-400	CMM	28-31-03	4	9/15/2006	CMM60-72101
			SB	60-72101-28-2	1	2/14/2008	60-72101-28-2
				<i>To convert the motor-impeller subassemblies to 60-72101-5 configuration.</i>			
			SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01
				<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>			
			SB	SB60-721-28-4	0	12/22/1998	SB60-721-28-4
				<i>Replacement of Thrust Washer P/N 60-06561</i>			
			SB	SB60-721-28-5	3	2/3/2004	SB60-721-28-5
				<i>Replaces inlet adapter to prevent excessive wear. Rev 1 adds pricing. Rev 3 updates pricing info for kits.</i>			
			TR	TR28-31-03-10	0	6/11/2009	TR28-31-03-10
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Page 7004, Table 7002.</i>			
			TR	TR28-31-03-11	0	6/11/2009	TR28-31-03-11
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Page 7014, Paragraph 4.C, Steps (7) and (8).</i>			
			TR	TR28-31-03-12	0	6/11/2009	TR28-31-03-12
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Pages 9007 and 9008, Table 9002.</i>			
			TR	TR28-31-03-13	0	6/11/2009	TR28-31-03-13
				<i>Revises Figure 3001, Sheet 2, Figure 3001, Sheet 3, Paragraph 3.B., Figure 7001, Sheet 4, Figure 7009, Figure 7012, Figure 7013</i>			
			TR	TR28-31-03-14	0	6/11/2009	TR28-31-03-14
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Page 6008, Paragraph 5.B CAUTION.</i>			
			TR	TR28-31-03-6	0	6/11/2009	TR28-31-03-6
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Page 1001, Step 1.B.</i>			
			TR	TR28-31-03-7	0	6/11/2009	TR28-31-03-7
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Page 1004, Step 4.A.</i>			
			TR	TR28-31-03-8	0	6/11/2009	TR28-31-03-8
				<i>This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Page 3007, Paragraph 4.O, Step (4).</i>			

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-72101-2, -3B, -4, -5	FUEL JETTISON PUMP MOTOR-IMPELLER SUBASSEMBLY	B-747, 747-400	TR	TR28-31-03-9	0	6/11/2009	TR28-31-03-9

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*This Temporary Revision (TR) revises CMM 28-31-03, Revision 4, Pages 7037 through 7044, Paragraph 4.E, Steps (1) through (12).*

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-755 SERIES	FUEL BOOST PUMP ASSEMBLY	B-747/757/737-700/800	CMM	28-22-06	7	5/25/2007	CMM60-755		
			SB	60-75501-28-1	0	4/16/2002	60-75501-28-1		
			<i>Documents modifications required to convert the fuel boost pump motor-impeller subassembly P/N 60-75501 to the improved 60-75501-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-75501-28-2	0	12/21/2005	60-75501-28-2		
			<i>To convert the motor-impeller subassemblies to 60-75501-5 configuration.</i>						
			SB	60-75503-28-1	0	4/16/2002	60-75503-28-1		
			<i>Documents modifications required to convert the fuel boost pump motor-impeller subassembly P/N 60-75503 to the improved 60-75503-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-75503-28-2	0	12/21/2005	60-75503-28-2		
			<i>To convert the motor-impeller subassemblies to 60-75503-5 configuration.</i>						
			SB	60-75504-28-1	0	4/16/2002	60-75504-28-1		
			<i>Documents modifications required to convert the fuel boost pump motor-impeller subassembly P/N 60-75504 to the improved 60-755100-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-75504-28-2	1	5/11/2006	60-75504-28-2		
<i>To convert the motor-impeller subassembly 60-75504 to the 60-755100-5 configuration.</i>									
SB	60-755100-28-1	0	4/16/2002	60-755100-28-1					
<i>Documents mods required to convert the fuel boost pump motor-impeller subassy P/N 60-755100 &amp; 60-755100-2 to the improved 60-755100-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>									
SB	60-755-28-2	0	7/1/1971	60-755-28-2					
<i>Improve cooling of bearing and thermal fuses and eliminate a potential air pocket in motor cavity.</i>									
SB	60-755-28-3	0	1/14/1972	60-755-28-3					
<i>Eliminate slightly greater wear on priming section side plates.</i>									
SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01					
<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>									



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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-755 SERIES	FUEL BOOST PUMP ASSEMBLY	B-747/757/737-700/800	SL	60-703-2-12	0	3/1/1999	60-703-2-12
			<i>(Description not available)</i>				
			TR	TR28-22-06-03	0	2/22/2017	TR28-22-06-03
			<i>Adds new part to IPL Figure 2, Item 185.</i>				

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-755100-2,-4,-5	FUEL BOOST PUMP MOTOR- IMPELLER SUBASSEMBLY	B-747/757/737-600, - 700, -800 757	CMM	28-22-08	5	6/16/2006	CMM60-755100		
			SB	60-75501-28-1	0	4/16/2002	60-75501-28-1		
			<i>Documents modifications required to convert the fuel boost pump motor-impeller subassembly P/N 60-75501 to the improved 60-75501-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-75503-28-1	0	4/16/2002	60-75503-28-1		
			<i>Documents modifications required to convert the fuel boost pump motor-impeller subassembly P/N 60-75503 to the improved 60-75503-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-75504-28-1	0	4/16/2002	60-75504-28-1		
			<i>Documents modifications required to convert the fuel boost pump motor-impeller subassembly P/N 60-75504 to the improved 60-755100-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-755100-28-1	0	4/16/2002	60-755100-28-1		
			<i>Documents mods required to convert the fuel boost pump motor-impeller subassy P/N 60-755100 &amp; 60-755100-2 to the improved 60-755100-4 config., which incorporates a new pottingless connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-755100-28-2	1	5/5/2006	60-755100-28-2		
<i>To convert the motor-impeller subassemblies 60-755100, -2, and -4 to 60-755100-5 configuration.</i>									
SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01					
<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>									
SL	SIL28-22-Aeroshell	0	12/10/2007	SIL28-22-Aeroshell					
<i>Announcement of BMS 3-33 (Aeroshell Grease 33) as substitute for Aeroshell Grease 16</i>									
SL	SIL28-22-Epocast	0	4/25/2012	SIL28-22-Epocast					
<i>Announce discontinuance of support for legacy motor-impeller subassemblies and cables which use Epocast 220A potting compound.</i>									
TR	TR28-22-08-1	0	6/25/2009	TR28-22-08-1					
<i>(Description not available)</i>									
TR	TR28-22-08-10	0	6/25/2009	TR28-22-08-10					
<i>(Description not available)</i>									

PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-755100-2,-4,-5	FUEL BOOST PUMP MOTOR- IMPELLER SUBASSEMBLY	B-747/757/737-600, - 700, -800 757	TR	TR28-22-08-11	0	6/25/2009	TR28-22-08-11
			<i>(Description not available)</i>				
			TR	TR28-22-08-12	0	6/25/2009	TR28-22-08-12
			<i>(Description not available)</i>				
			TR	TR28-22-08-13	0	6/25/2009	TR28-22-08-13
			<i>(Description not available)</i>				
			TR	TR28-22-08-14	0	6/25/2009	TR28-22-08-14
			<i>(Description not available)</i>				
			TR	TR28-22-08-15	0	6/25/2009	TR28-22-08-15
			<i>(Description not available)</i>				
			TR	TR28-22-08-16	0	6/25/2009	TR28-22-08-16
			<i>(Description not available)</i>				
			TR	TR28-22-08-17	0	6/25/2009	TR28-22-08-17
			<i>(Description not available)</i>				
			TR	TR28-22-08-18	0	6/25/2009	TR28-22-08-18
			<i>(Description not available)</i>				
TR	TR28-22-08-19	0	5/25/2011	TR28-22-08-19			
<i>(Description not available)</i>							
TR	TR28-22-08-2	0	6/25/2009	TR28-22-08-2			
<i>(Description not available)</i>							
TR	TR28-22-08-20	0	5/25/2011	TR28-22-08-20			
<i>(Description not available)</i>							
TR	TR28-22-08-21	0	5/25/2011	TR28-22-08-21			
<i>(Description not available)</i>							
TR	TR28-22-08-22	0	5/25/2011	TR28-22-08-22			
<i>(Description not available)</i>							
TR	TR28-22-08-23	0	2/15/2012	TR28-22-08-23			
<i>(Description not available)</i>							
TR	TR28-22-08-24	0	2/15/2012	TR28-22-08-24			
<i>(Description not available)</i>							

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-755100-2,-4,-5	FUEL BOOST PUMP MOTOR- IMPELLER SUBASSEMBLY	B-747/757/737-600, - 700, -800 757	TR	TR28-22-08-25	0	2/15/2012	TR28-22-08-25		
			<i>(Description not available)</i>						
			TR	TR28-22-08-26	0	5/11/2022	TR28-22-08-26		
			<i>(Description not available)</i>						
			TR	TR28-22-08-27	0	2/22/2017	TR28-22-08-27		
			<i>(Description not available)</i>						
			TR	TR28-22-08-3	0	6/25/2009	TR28-22-08-3		
			<i>(Description not available)</i>						
			TR	TR28-22-08-4	0	6/25/2009	TR28-22-08-4		
			<i>(Description not available)</i>						
TR	TR28-22-08-5	0	6/25/2009	TR28-22-08-5					
<i>(Description not available)</i>									
TR	TR28-22-08-6	0	6/25/2009	TR28-22-08-6					
<i>(Description not available)</i>									
TR	TR28-22-08-7	0	6/25/2009	TR28-22-08-7					
<i>(Description not available)</i>									
TR	TR28-22-08-8	0	6/25/2009	TR28-22-08-8					
<i>(Description not available)</i>									
TR	TR28-22-08-9	0	6/25/2009	TR28-22-08-9					
<i>(Description not available)</i>									



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-843, -1, -2, -2M, -3, -4; 60-845, -1, -2, -2M, -3, -4	FUEL BOOST PUMP HOUSING SUBASSEMBLY	DC-10, KC-10, KDC- 10, MD-10, MD-11	CMM	28-22-01	6	10/23/2014	CMM60-843_60-845
			SB	28-5	0	2/15/1973	28-5
				<i>Provide an improved connector receptacle having increased individual pin/ socket forces and modified interfacial seal provisions.</i>			
			SB	60-843_845-28-1	0	7/27/2005	60-843_845-28-1
				<i>Provides instructions to install a new connector assembly. Product improvement designed to prevent accumulation of corrosion or conductive contaminants which can lead to electrical arcing.</i>			
			SB	60-843_845-28-2	0	10/1/2014	60-843_845-28-2
				<i>This service bulletin provides instructions to install a new connector assembly, P/N 60-843504 and rolls the configuration dash numbers.</i>			
			SB	60-843_845-28-22	0	9/30/1980	60-843_845-28-22
				<i>Some end caps not within design tolerance may allow fuel leakage.</i>			
			SB	60-843-3-28-14	1	10/22/1981	60-843-3-28-14
				<i>Prevent installation of pumping element into an incorrectly converted fuel booster pump assembly and provides an ablative electrical surface. New protector cap enables operator to replace connector assembly without removing pump housing from aircraft.</i>			
			SIL	SIL 60-843-845-3-28-2	0	1/18/2010	SIL 60-843_845-3-28-2
			SIL	SIL60-843-845-3-28-1	0	2/22/2007	SIL60-843-845-3-28-1
			SL	60-843_845-4-28-1	0	10/15/2015	60-843_845-4-28-1
				<i>(Description not available)</i>			



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-847-1A, 60-847-2, 60-847-3, 60-847-4	FUEL BOOST PUMP ASSEMBLY	DC-10, MD-11	CMM	28-22-03	9	10/15/2013	CMM60-847		
			SB	60-847-1A-28-6	0	2/15/1973	60-847-1A-28-6		
			<i>Change reprime provisions resulting from DACO fuel system changes.</i>						
			SB	60-847-28-1	0	6/4/1999	60-847-28-1		
			<i>Replacement of thrust washer P/N 60-06561 with P/N 60- 05118.</i>						
			SB	60-847-28-2	0	12/2/2002	60-847-28-2		
			<i>Provides supplemental inspection instructions to ensure proper routing of lead wires connecting the pumping unit stator to the pump connector. Supplement to procedures provided in CMM ATA 28-22-03.</i>						
			SB	60-847-28-3	2	5/16/2008	60-847-28-3		
			<i>To convert fuel boost pump pumping units 60-847-1A,-2,-3 to 60-847-4 configuration.</i>						
			SB	60-847-3-28-13	0	3/17/1975	60-847-3-28-13		
<i>Provide a procedure for salvage of pump housing or connector support, and provide an improved bearing-to-housing wear interface.</i>									
SIL	60-847-8-203-1	0	8/9/1972	60-847-8-203-1					
SIL	SIL60-847-28-2	0	8/7/2008	SIL60-847-28-2					
SIL	SIL60-847-28-3	0	10/10/2013	SIL60-847-28-3					
SIL	SIL60-847-28-4	0	6/8/2012	SIL60-847-28-4					
SIL	SIL60-847-28-5	0	5/15/2015	SIL60-847-28-5					



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-989, -1, -2, -3, -5, -6, -8	FUEL BOOST PUMP ASSEMBLY	B-747 (JT9D-70, -70)	CMM	28-22-04	5	1/9/2004	CMM60-989
			SB	60-98902-28-1	0	4/16/2002	60-98902-28-1
			<i>Documents mods required to convert the P/N 60-98902, 60-98902A Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98902-28-2	0	12/21/2005	60-98902-28-2
			<i>To convert the motor-impeller subassemblies to 60-98976-5 configuration.</i>				
			SB	60-98904-28-1	0	4/16/2002	60-98904-28-1
			<i>Documents mods required to convert the P/N 60-98904, 60-98904A Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98904-28-2	0	12/21/2005	60-98904-28-2
			<i>To convert the motor-impeller subassemblies to 60-98976-5 configuration.</i>				
			SB	60-98906-28-1	0	4/16/2002	60-98906-28-1
			<i>Documents mods required to convert the P/N 60-98906, 60-98906A Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98906-28-2	0	12/21/2005	60-98906-28-2
			<i>To convert the motor-impeller subassemblies 60-98906 and 60-98906A to the 60-98976-5 configuration</i>				
			SB	60-989-28-18	1	3/6/1980	60-989-28-18
			<i>Prevent wear caused by mating inlet check valve poppet.</i>				
			SB	60-989-28-19	2	4/25/1988	60-989-28-19
			<i>Prevent interchangeability of low flow pump at this location.</i>				
			SB	60-989-28-20	1	8/15/1993	60-989-28-20
			<i>Modify to seal cable assembly.</i>				
			SB	60-98976-28-1	0	4/16/2002	60-98976-28-1
			<i>Documents mods required to convert the P/N 60-98976, 60-98976-1 Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98976-28-3	0	12/21/2005	60-98976-28-3
			<i>To convert the motor-impeller subassemblies to 60-98976-5 configuration.</i>				

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-989, -1, -2, -3, -5, -6, -8	FUEL BOOST PUMP ASSEMBLY	B-747 (JT9D-70, -70)	SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01
<p><i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i></p>							
60-98901, 60-98901A, 60-98903, 60-98905	FUEL BOOST PUMP HOUSING SUBASSEMBLY	747	CMM	28-22-11	1	1/11/2018	CMM60-98901

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-989100-2, -4, -5	FUEL OVERRIDE PUMP MOTOR-IMPELLER SUBASSEMBLY	B-757/737-700/737-800	CMM	28-22-09	4	4/28/2006	CMM60-989100		
			SB	60-98908-28-1	0	4/16/2002	60-98908-28-1		
			<i>Documents mods required to convert the P/N 60-98908, 60-989100 Fuel Pump Motor-Impeller Subassemblies to the improved 60-989100-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
			SB	60-989100-28-1	0	4/16/2002	60-989100-28-1		
			<i>Modification of Motor-Impeller Subassembly, p/n 60-989100-2 (Boeing p/n 60B89004-12) to p/n 60-989100-4 (Boeing p/n 60B89004-14).</i>						
			SB	60-989100-28-2	0	5/25/2005	60-989100-28-2		
			<i>To convert the motor-impeller subassemblies 60-989100-2 and 60-989100-4 to the 60-989100-5 configuration.</i>						
			SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01		
			<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>						
			TR	TR28-22-09-10	0	6/25/2009	TR28-22-09-10		
			<i>(Description not available)</i>						
			TR	TR28-22-09-11	0	6/25/2009	TR28-22-09-11		
			<i>(Description not available)</i>						
TR	TR28-22-09-12	0	6/25/2009	TR28-22-09-12					
<i>(Description not available)</i>									
TR	TR28-22-09-13	0	6/25/2009	TR28-22-09-13					
<i>(Description not available)</i>									
TR	TR28-22-09-14	0	6/25/2009	TR28-22-09-14					
<i>(Description not available)</i>									
TR	TR28-22-09-15	0	6/25/2009	TR28-22-09-15					
<i>(Description not available)</i>									
TR	TR28-22-09-16	0	6/25/2009	TR28-22-09-16					
<i>(Description not available)</i>									

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PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-989100-2, -4, -5	FUEL OVERRIDE PUMP MOTOR-IMPELLER SUBASSEMBLY	B-757/737-700/737-800	TR	TR28-22-09-17	0	6/25/2009	TR28-22-09-17		
			<i>(Description not available)</i>						
			TR	TR28-22-09-18	0	8/15/2011	TR28-22-09-18		
			<i>(Description not available)</i>						
			TR	TR28-22-09-19	0	8/15/2011	TR28-22-09-19		
			<i>(Description not available)</i>						
			TR	TR28-22-09-20	0	2/13/2012	TR28-22-09-20		
			<i>(Description not available)</i>						
			TR	TR28-22-09-21	0	2/13/2012	TR28-22-09-21		
			<i>(Description not available)</i>						
			TR	TR28-22-09-22	0	2/13/2012	TR28-22-09-22		
			<i>(Description not available)</i>						
			TR	TR28-22-09-23	0	12/11/2017	TR28-22-09-23		
			<i>(Description not available)</i>						
			TR	TR28-22-09-24	0	12/11/2017	TR28-22-09-24		
			<i>(Description not available)</i>						
TR	TR28-22-09-25	0	12/11/2017	TR28-22-09-25					
<i>(Description not available)</i>									
TR	TR28-22-09-26	0	12/11/2017	TR28-22-09-26					
<i>(Description not available)</i>									
TR	TR28-22-09-3	0	6/25/2009	TR28-22-09-3					
<i>(Description not available)</i>									
TR	TR28-22-09-4	0	6/25/2009	TR28-22-09-4					
<i>(Description not available)</i>									
TR	TR28-22-09-5	0	6/25/2009	TR28-22-09-5					
<i>(Description not available)</i>									
TR	TR28-22-09-6	0	6/25/2009	TR28-22-09-6					
<i>(Description not available)</i>									

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60-989100-2, -4, -5	FUEL OVERRIDE PUMP MOTOR-IMPELLER SUBASSEMBLY	B-757/737-700/737- 800	TR	TR28-22-09-7	0	6/25/2009	TR28-22-09-7		
			<i>(Description not available)</i>						
			TR	TR28-22-09-8	0	6/25/2009	TR28-22-09-8		
<i>(Description not available)</i>									
60-989-4, -7, 60-98908, 60-989100, 60-98907	FUEL OVERRIDE PUMP ASSEMBLY, MOTOR- IMPELLER SUBASSEMBLY, HOUSING SUBASSEMBLY	B-757/737-700/800	CMM	28-22-05	5	6/22/2006	CMM60-989-4		
			SB	60-98908-28-1	0	4/16/2002	60-98908-28-1		
			<i>Documents mods required to convert the P/N 60-98908, 60-989100 Fuel Pump Motor-Impeller Subassemblies to the improved 60-989100-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>						
SB	60-98908-28-2	0	5/25/2005	60-98908-28-2					
<i>To convert the motor-impeller subassemblies 60-98908 and 60-989100 to the 60-989100-5 configuration.</i>									
SB	Crane_HA_Motor_Imp-28- 01	2	3/21/2003	Crane_HA_Motor_Imp-28- 01					
<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>									



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.
60-98976-2, -4, -5	FUEL BOOST PUMP MOTOR- IMPELLER SUBASSEMBLY	B-747	CMM	28-22-07	2	5/25/2007	CMM60-98976
			SB	60-98902-28-1	0	4/16/2002	60-98902-28-1
			<i>Documents mods required to convert the P/N 60-98902, 60-98902A Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98904-28-1	0	4/16/2002	60-98904-28-1
			<i>Documents mods required to convert the P/N 60-98904, 60-98904A Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98906-28-1	0	4/16/2002	60-98906-28-1
			<i>Documents mods required to convert the P/N 60-98906, 60-98906A Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98976-28-1	0	4/16/2002	60-98976-28-1
			<i>Documents mods required to convert the P/N 60-98976, 60-98976-1 Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new end cap and connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98976-28-2	0	4/16/2002	60-98976-28-2
			<i>Documents mods required to convert the P/N 60-98976-2 Fuel Pump Motor-Impeller Subassemblies to the improved 60-98976-4 config., which incorporates a new connector. Prevents accumulation of corrosion/conductive fuel contaminants.</i>				
			SB	60-98976-28-4	1	3/15/2008	60-98976-28-4
			<i>To convert the motor-impeller subassemblies to 60-98976-5 configuration.</i>				
			SB	Crane_HA_Motor_Imp-28-01	2	3/21/2003	Crane_HA_Motor_Imp-28-01
			<i>Provides supplemental inspection and assembly instructions to ensure proper routing of wire bundle connecting motor-impeller stator to the pump connector. Procedures supplement assembly procedures provided in CMMs given in paragraph 1.L.</i>				
			TR	TR28-22-07-10	0	6/11/2009	TR28-22-07-10
			<i>Revises pages 9007 and 9008, Table 9002.</i>				



PART NUMBER	PART NAME	AIRCRAFT MODEL	DOCUMENT TYPE	ATA NUMBER	REVISION	DATE	DOCUMENT NO.		
60-98976-2, -4, -5	FUEL BOOST PUMP MOTOR- IMPELLER SUBASSEMBLY	B-747	TR	TR28-22-07-11	0	6/11/2009	TR28-22-07-11		
				<i>Revises (1) Page 3009, Figure 3001, Sheet 2 (2 Places), (2) Page 3010, Figure 3001, Sheet 3 (2 Places), (3) Page 7003, Paragraph 3.B, (4) Page 7008, Figure 7001, Sheet 4 (2 Places), (5) Page 7026, Figure 7006 (2 Places), (6) Page 7038, Figure 7010 (2 Places), (7) Page 7039, Figure 7011 (2 Places), (8) Page 7040, Figure 7012.</i>					
			TR	TR28-22-07-12	0	6/11/2009	TR28-22-07-12		
				<i>Revises page 6007, Paragraph 5.B CAUTION.</i>					
			TR	TR28-22-07-18	0	3/1/2021	TR28-22-07-18		
				<i>Revises page 10020, IPL Figure 1, Item 180 and adds Item -180A.</i>					
			TR	TR28-22-07-19	0	3/1/2021	TR28-22-07-19		
				<i>Revises page 10007, Table 10002.</i>					
			TR	TR28-22-07-3	0	6/11/2009	TR28-22-07-3		
				<i>Revises page 1001, Step 1.B.</i>					
TR	TR28-22-07-4	0	6/11/2009	TR28-22-07-4					
	<i>Revises page 1004, Step 4.A.</i>								
TR	TR28-22-07-5	0	6/11/2009	TR28-22-07-5					
	<i>Delete Page 3006, Paragraph 4.Q, Step (4) CAUTION</i>								
TR	TR28-22-07-6	0	6/11/2009	TR28-22-07-6					
	<i>Rearrange Paragraph 4.D, Steps (4) through (8)</i>								
TR	TR28-22-07-7	0	6/11/2009	TR28-22-07-7					
	<i>Delete Page 7042, Paragraph 4.E, Step (10).</i>								
TR	TR28-22-07-8	0	6/11/2009	TR28-22-07-8					
	<i>Revises Table 7002.</i>								
TR	TR28-22-07-9	0	6/11/2009	TR28-22-07-9					
	<i>Revises page 7010, Paragraph 4.B, Steps (7) and (8).</i>								
68-119A	TBD	VERTOL DYN.AIR, V234 Helicopter	CMM	29-10-02	0	11/15/1982	CMM68-119A		
70146	FUEL BOOSTER PUMP	BEECH-120	OHM	28-20-07	1	9/1/1968	OHM70146		
70379	TBD	VARIOUS	CMM	29-20-70	0	10/23/2009	CMM70379		

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70380	TBD	?	TM	(No ATA Number)	0	12/4/1969	TM70380
70408-2	HYDRAULIC SHUTTLE VALVE	B-727	OHM	29-20-68	2	11/20/1990	OHM70408-2
70566-3	ROTOR BRAKE	Various	CMM	62-40-01	0	5/20/2012	CMM70566-3
			TR	TR62-40-01-01	0	12/18/2013	TR62-40-01-01
<i>(Description not available)</i>							
70581	DISCONNECT SHUTTLE VALVE	B-727	OHM	32-40-01	1	2/14/1963	OHM70581
70623-2A	FUEL BOOSTER PUMP	BEECH-65	OHM	28-20-26	1	9/22/1986	OHM70623-2A
70623,-1,-2,-3,-3A	FUEL BOOSTER PUMPS	MERLIN 2A & 2B	OHM	28-20-02	0	4/15/1969	OHM70623
71126, -1, -3,-7,-9,-15,-17, -19	FUEL BOOSTER PUMP	BEECH-90,-15; METROLINER; BAe JETSTREAM 31	OHM	28-20-03	2	6/30/1997	OHM71126-1
71154, -1, 72443	FUEL BOOSTER PUMP	BEECH-80, -90, -100, -115	OHM	28-20-01	1	12/15/1972	OHM71154
71234	HYDRAULIC HAND PUMP ASSEMBLY	B-727/737/747/757/767/777	CMM	29-20-65	10	6/9/2021	CMM71234
71314	FUEL BOOSTER PUMP	HFB-320	OHM	28-20-04	1	7/2/1981	OHM71314
71383-2	FUEL BOOSTER PUMP	DHC-6	OHM	28-20-29	1	2/27/1986	OHM71383-2
71383-3, -3A	FUEL BOOSTER PUMP	DHC-6	OHM	28-20-06	1	3/1/1969	OHM71383-3
71420-1, -2, -3, -4, -5, -6, -7, -8	3-WAY, 2-POSITION SOLENOID-OPERATED VALVE ASSEMBLY	Gulfstream	CMM	29-20-71	0	10/1/2008	CMM71420
71477-3,-4	DISCONNECT SHUTTLE VALVE	B-707/727	OHM	32-40-05	0	1/5/1980	OHM71477-3

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71494	PROPELLER BRAKE CONTROL VALVE	CV-580	OHM	(No ATA Number)	0	8/1/1973	OHM71494
71538, -1, -2	DIRECTIONAL CONTROL VALVE	F-28	OHM	29-20-23	3	2/28/1992	71538
			SB	71538-29-1	0	2/15/1992	71538-29-1
			<i>Increases lever travel between internal stops and decreases tolerance band to ease adjustment at installation.</i>				
71539, 71539-1	DIRECTIONAL CONTROL VALVE	F-28/FOKKER 100	OHM	29-20-24	3	3/1/1996	OHM71539
			SB	71539-29-1	0	7/21/1995	71539-29-1
			<i>To ensure gear door retracts.</i>				
71540, 71540-1, 71541, 71541-1	DIRECTIONAL CONTROL VALVE	F-28/FOKKER 100	OHM	29-20-25	0	9/30/2010	CMM71540
			SB	71541-29-1	0	2/28/1992	71541-29-1
			<i>Makes valve less sensitive to fluid contamination.</i>				
71542, 72428, -1	HYDRAULIC SHUTTLE VALVE	F-28/F-228/FOKKER 100	CMM	29-20-26	2	1/15/1992	CMM71542
			SB	72428-29-1	1	2/15/1994	72428-29-1
<i>Allow modification of valve from flared to flareless fittings or from flareless to flared fittings.</i>							
71543	TBD	F-28/F-228	OHM	29-20-27	1	9/1/1991	OHM71543
71544, -1	DIRECTIONAL CONTROL VALVE	F-28/FOKKER 100	OHM	29-20-28	3	9/1/1991	OHM71544
71545,-1	TWO-POSITION, THREE-WAY MECHANICALLY OPERATED DUMP VALVE	F-28/FOKKER 100	OHM	29-20-29	2	4/15/1992	OHM71545

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71577-2,-3,-4,-5	VALVE DISCONNECT	B-737-200/747	CMM	32-40-08	5	9/1/2016	CMM71577
			SB	71577-2-32-167	0	10/9/1987	71577-2-32-167
			<i>Stress corrosion failures have occurred with bolts P/N 83443-2 which are made of H-II steel.</i>				
71577-2,-3,-4,-5	VALVE DISCONNECT	B-737-200/747	SB	71577-3-32-172	0	2/8/1988	71577-3-32-172
			<i>Design improvements to prevent stress corrosion failures.</i>				
71580-7,-8	THREE WAY PILOT OPERATED VALVE ASSEMBLY	F-28/FOKKER 100	OHM	29-20-31	1	9/1/1991	OHM71580-7
71599,-1	FUEL BOOSTER PUMP	BEECH-65, -80, -115	OHM	28-20-05	1	5/1/1977	OHM71599
71708-3,-4,-8, -9	FUEL BOOSTER PUMP	BEECH-56, -60	OHM	28-20-21	0	10/15/1970	OHM71708-3
71708-6	FUEL BOOSTER PUMP	BEECH-60	OHM	28-20-13	1	1/1/1969	OHM71708-6
71727	TWO-WAY RELIEF VALVW	DC-8, DC-9	OHM	29-20-10	0	8/1/1966	OHM71727
71742-1, -2	DISCONNECT SHUTTLE VALVE ASSEMBLY	B-707/727	CMM	32-40-83	4	12/15/2007	OHM71742-1
71852-1	HYDRAULIC SELECTOR VALVE	MAGNAGHI	OHM	29-20-54	0	1/1/1971	OHM71852-1
71863-1, -2, -3, -4, -5, -6, -7, -8	3-WAY, 2-POSITION, SOLENOID OPERATED, HYDRAULIC VALVE	VARIOUS	CMM	32-48-22	0	11/30/2011	CMM71863
71985-2	PRESSURE RELIEF CARTRIDGE	B-747	OHM	29-20-36	2	2/12/1988	OHM71985-2
71998, 72926-1, -2	HYDRAULIC SHUTTLE VALVE	B-737/747	SB	29-1	0	2/28/1972	29-1
<i>Add a bias feature to favor A hydraulic system source.</i>							
72022	HYDRAULIC RESERVOIR AIR PRESSURE REGULATOR CARTRIDGE VALVE	B-747	OHM	29-20-37	1	6/15/1974	OHM72022

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72086	HYDRAULIC RELIEF VALVE	B-747	OHM	29-20-38	2	10/15/1990	OHM72086
72266-3, 72390-3	FUEL BOOSTER CANISTER PUMP	BEECH, DHC-6	CMM	28-20-24	1	2/21/1986	OHM72266
72271	FUEL BOOSTER PUMP	BEECH 100	OHM	28-20-16	1	8/8/1984	OHM72271
72285-1	FUEL BOOSTER PUMP	M-226, MERLIN-3	OHM	28-20-20	0	1/1/1970	OHM72285-1
72286	FUEL BOOSTER PUMP	M-226/226TC	OHM	28-20-19	0	11/15/1969	OHM72286
72368-1	FUEL BOOSTER PUMP	DHC-6	OHM	28-20-22	2	1/21/1983	OHM72368-1
72385-2	3-WAY, 2-POSITION SOLENOID OPERATED HYDRAULIC VALVE ASSEMBLY	VARIOUS	CMM	29-20-77	0	5/24/2021	72385-2
72390-3, -3B, -4, 4A, 72392	FUEL BOOST CANISTER PUMP	BEECH,EMBRAER	OHM	28-20-23	1	4/10/1979	OHM72390-3
			SB	72390-3-28-15	2	7/15/1977	72390-3-28-15
			<i>Design modification to increase life expectancy of unit.</i>				
			SB	72390-3-28-17	0	6/1/1978	72390-3-28-17
			<i>Fuel Boostere Pump, Modification from p/n 72390-3 to p/n 72390-3B</i>				
			SB	72390-4-28-16	1	6/15/1978	72390-4-28-16
			<i>Eliminates minor pressure fluctuations of sporadic nature which occasionally cause nuisance low pressure cockpit annunciation.</i>				
			SB	72390-4A-28-21	2	6/8/1982	72390-4A-28-21
			<i>Increase operational life of unit.</i>				
72407, -1, 72408, -1	CARTRIDGE-TYPE RELIEF VALVE	F-28/FOKKER 100	CMM	29-20-57	2	4/30/1992	OHM72407
72427,-1,-3	DUAL HYDRAULIC POWER BRAKE VALVE	FOKKER F-28	OHM	29-20-30	1	9/26/1980	OHM72427
72444	FUEL BOOSTER PUMP	M-226/226TC	OHM	28-20-17	0	7/15/1978	OHM72444

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72477	FUEL BOOSTER PUMP	M-2A/2B	OHM	28-20-18	0	3/15/1975	OHM72477
72484	HYDRAULIC PRIORITY VALVE	B-707/727	OHM	29-20-58	0	1/15/1975	OHM72484
			SB	72484-29-4	0	6/1/1977	72484-29-4
			<i>Design modification to provide a stronger end closure.</i>				
72491-1	HYDRAULIC RELIEF VALVE ASSEMBLY	VARIOUS	CMM	29-20-76	0	4/7/2021	72491-1
72515-2,-3, -4, 72516-3, -4, -5, -6, -7	SELENOID BRAKE SUPPLY VALVE	L-1011	OHM	32-43-08	0	8/29/1980	OHM72515-2
			SB	72515_72516-29-5	0	7/15/1977	72515_72516-29-5
			<i>Provides greater longevity to seal at all flow rates, improves valve maintainability improves seal compound material, reduces cost of replacement seal and improves dispatch reliability.</i>				
			SB	72515_72516-29-7	0	12/18/1979	72515_72516-29-7
			<i>Material changes to reduce internal leakage, wear and prevent incorrect installation in aircraft.</i>				
			SB	72516-29-6	0	9/30/1978	72516-29-6
			<i>Prevents incorrect aircraft installation by replacement of electrical connector.</i>				
72532-1,-2,-3	PRESSURE CONTROLLED HYDRAULIC CHECK VALVE	L-1011	OHM	32-33-04	1	1/25/1983	OHM72532-1
			SB	72532-1-32-86	0	4/9/1981	72532-1-32-86
			<i>Replaces aluminum spring guide with steel part and modifies poppet 85108.</i>				
			SB	72532-2-32-100	0	10/29/1982	72532-2-32-100
			<i>Replace integral relief spring with new spring which will operate at a lower stress level to eliminate possible spring breakage and increase operational life.</i>				
72553-1, -2	NLG RETRACTION ACTUATOR HYDRAULIC SHUTTLE VALVE	L-1011	OHM	32-34-10	0	7/11/1980	OHM72553-1
72554-3,-4,-5	MLG RETRACTION ACTUATOR RELIEF VALVE	L-1011	OHM	32-32-04	0	7/15/1980	OHM72554-3

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72555-1, -2	NLG RETRACTION ACTUATOR RELIEF VALVE	L-1011	OHM	32-34-02	0	11/1/1972	OHM72555-1
72556-1	MLG BOGIE POSITIONER HYDRAULIC VALVE	L-1011	OHM	32-31-13	0	7/1/1978	OHM72556-1
			SL	72556-1-9-8	0	9/16/1986	72556-1-9-8
			<i>(Description not available)</i>				
72564	VACUUM-CONTROLLED HYDRAULIC CHECK VALVE	L-1011	OHM	32-31-11	1	1/1/1971	OHM72564
72596	TBD	Various	CMM	28-20-36	0	1/1/1901	CMM72596
72601-7	VALVE PRESSURE REGULATOR MODULE ASSEMBLY	L-1011	OHM	27-21-17	2	12/17/1982	OHM72601-7
72665-1	HYDRAULIC SELECTOR VALVE MODULE	L-1011	OHM	32-71-02	0	5/15/1979	OHM72665-1
72744	BALANCED RELIEF AND BYPASS VALVE	DC-10	CMM	29-12-11	0	5/1/1971	CMM72744
72848	HYDRAULIC HAND PUMP	DC-10	CMM	29-12-13	0	4/15/1976	CMM72848
72854	BALANCED RELIEF AND BYPASS VALVE	DC-10	CMM	28-12-10	0	3/1/1972	CMM72854
72887, -1, -2, -3	TBD	Various	CMM	29-20-74	0	1/1/1901	CMM72887

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72909	HYDRAULIC RESERVOIR PRESSURE REGULATOR	B-707/727/737-200	CMM	29-01-03	3	11/15/2007	CMM72909	
			SB	72909-29-2	2	12/1/1975	72909-29-2	
			<i>Increase life of poppet and valve seat by reducing wear on valve seat during operation.</i>					
			SB	72909-29-28	0	2/22/1984	72909-29-28	
			SB	72909-29-28	0	2/22/1984	72909-29-28	
<i>Replace square valve seat with round valve.</i>								
			SL	72909-12-15	0	12/15/1989	72909-12-15	
<i>(Description not available)</i>								
72912	NLG RETRACTION ACTUATOR HYDRAULIC VALVE	L-1011	OHM	32-34-11	0	4/1/1973	OHM72912	
72926-1, -2	HYDRAULIC SHUTTLE VALVE	VARIOUS	CMM	29-20-73	0	10/8/2011	72926	
77-389,-3	TRANSDUCER-MOTOR ASSEMBLY	B-747SR	OHM	32-42-48	1	11/28/1984	OHM77-389	
			SB	77-389-32-66	0	12/17/1979	77-389-32-66	
			<i>Prevents possible interference between coil and rotating components.</i>					
			SB	77-389-32-85	0	1/30/1981	77-389-32-85	
			<i>Prevents deformation of spacer tube.</i>					
			SB	77-389-32-93	0	7/7/1981	77-389-32-93	
<i>Increases fan motor stall torque, especially at lower temperatures, without appreciable change in running temperature, starting current or power consumption.</i>								
			SB	77-389-32-96	1	4/2/1982	77-389-32-96	
<i>Changes bearings to improve service life.</i>								
77-395	TBD	B-767	CMM	32-42-71	2	6/1/1993	CMM77-395	
84-005	POWER BRAKE AND ANTI- SKID VALVE ASSEMBLY	PC-21	CMM	32-48-87	0	10/6/2022	CMM84-005	
84-017	TBD	EMB 314 (SUPER TACANO)	CMM	32-43-84	0	5/14/2010	CMM84-017	

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84-023, -01, -02	POWER BRAKE AND ANTISKID VALVE ASSEMBLY	HONDAJET	CMM	32-48-86	1	6/8/2018	CMM84-023
			SB	84-023-02-32-1	2	10/31/2019	84-023-02-32-1
			<i>To prevent O-Ring damage and internal leakage.</i>				
84-031, 37-451, 37-451-01	POWER BRAKE AND ANTISKID VALVE ASSEMBLY	PC-24	CMM	32-43-85	0	1/30/2020	CMM84-031
90-193, A, B	WATER-METHANOL REGULATOR VALVE ASSEMBLIES	CV-440, CV-580	OHM	73-20-01	0	5/1/1971	OHM90-193A
97-001A, B	VALVE TEST SET	MKII & MKIII	O&M	32-42-47	0	9/15/1978	O&M97-001A
98-027	ANTI-SKID SYSTEM TEST SET	NAMC	O&M	32-40-74	1	4/18/1986	O&M98-027
99-301	TBD	NAMC	O&M	(No ATA Number)	0	7/15/1971	O&M99-301
99-575	HYTROL MARK III CABLE SWITCH BOX ASSEMBLY	B-747	O&M	32-42-41	1	9/26/1983	O&M99-575
			SB	99-575-32-171	0	6/29/1988	99-575-32-171
			<i>Product improvement. Permits increased scope of testing autobrake control unit P/N 42-319-2.</i>				
			SB	99-575-32-38	0	6/15/1976	99-575-32-38
			<i>Eliminate EMI problem during performance of control box functional test.</i>				
			O&M	99-575-32-99	0	10/28/1982	99-575-32-99
			<i>To make bench test of 42-319-1, -3 autobrake control box compatible with actual aircraft test conditions with annunciator of DIM power.</i>				
99-577	ANTI-SKID CHECKOUG PROCEDURE	MKIII GA	O&M	(No ATA Number)	0	7/1/1977	O&M99-577
99-725	DC-9/DC-9-80 AUTOBRAKE CONTROLLER TEST SET	DC-9AB	OMM	(No ATA Number)	1	5/31/2008	OMM99-725
99-725-1	AUTOBRAKE CONTROL UNIT TEST SET	MD-80	O&M	(No ATA Number)	0	3/16/1990	O&M99-725-1

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99-777	DC-10 AUTOBRAKE CONTROLLER TEST SET	DC-10AB	OMM	(No ATA Number)	1	4/22/2008	OMM99-777
99-791	ANTI-SKID SYSTEM UNIVERSAL FIELD TESTER	MOST GENERAL AVIATION	OMM	(No ATA Number)	5	6/15/2008	OMM99-791
99-791-509	ADAPTER CABLE (CESSNA CITATION III (650))	CESSNA 650	OPER	(No ATA Number)	2	5/15/1996	OPER99-791-509
99-791-511	ADAPTER CABLE FOR CESSNA CITATION II (550 ADVANCED)	CESSNA 550ADV	OPER	(No ATA Number)	3	4/15/1991	OPER99-791-511
99-791-513	ADAPTER CABLE FOR CESSNA CITATION I, II (500, 550)	CESSNA 500/550	OPER	(No ATA Number)	2	4/15/1991	OPER99-791-513
99-791-515	ADAPTER CABLE FOR AERMACCHI MB-399	AERMACCHI MB339	OPER	(No ATA Number)	1	3/22/1991	OPER99-791-515
99-791-519	ADAPTER CABLE FOR EMBRAER EMB-120	EMB-120	OPER	(No ATA Number)	0	6/17/1985	OPER99-791-519
99-791-521	ADAPTER CABLE FOR SAAB FAIRCHILD SF340	SAAB 340	OPER	(No ATA Number)	0	2/1/1984	OPER99-791-521
99-791-531	ADAPTER CABLE FOR ATR 42	ATR-42	OPER	(No ATA Number)	0	10/23/1986	OPER99-791-531
99-791-563	TESTER ADAPTER CABLE	CESSNA 560	OPER	N/A	2	2/25/2001	OMM99-791-563
99-791-569	ADAPTER CABLE ASSEMBLY	CESSNA 750	OPER	(No ATA Number)	1	9/6/2006	OPER99-791-569
99-791-570-1	ADAPTER CABLE ASSEMBLY	RAYTHEON PREMIER 1A	OM	(No ATA Number)	1	10/13/2006	OM99-791-570-1
99-791-571	TESTER ADAPTER CABLE	CESSNA CITATION (560)	OPER	N/A	0	8/29/1997	OMM99-791-571
99-791-572	TESTER ADAPTER CABLE	CESSNA 560-BPC	OPER	N/A	1	3/31/2006	OMM99-791-572

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99-791-573	TESTER ADAPTER CABLE	CESSNA 525A	OPER	(No ATA Number)	0	9/30/2000	OPER99-791-573
99-791-577	ADAPTER CABLE ASSEMBLY	CESSNA CITATION MODEL 680 SOVEREIGN	OPER	(No ATA Number)	0	5/5/2005	OPER99-791-577
99-791-581	ADAPTER CABLE ASSEMBLY	CESSNA 525 CJ3	OM	(No ATA Number)	1	11/4/2005	OM99-791-581

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99-811	SEMIAUTOMATIC LRU TEST SET FOR BOEING 757/767	B-757/767	OMM	32-42-58	7	9/30/1997	OMM99-811		
			SB	99-811-32-10	0	8/9/2001	99-811-32-10		
			<i>Contains procedures for upgrading the test set software to version 7.0.</i>						
			SB	99-811-32-123	0	2/21/1983	99-811-32-123		
			<i>Revised limits on transducer interface.</i>						
			SB	99-811-32-129	0	6/30/1983	99-811-32-129		
			<i>Accommodate test parameters of new P/N 42-629-1 control unit.</i>						
			SB	99-811-32-131	0	11/23/1983	99-811-32-131		
			<i>Accommodate test parameters of new P/N 42-641-1 control unit.</i>						
			SB	99-811-32-134	0	1/1/1984	99-811-32-134		
			<i>Accommodate new LRU P/N 42-757 control unit and update wheel card fail driver test to latest configuration.</i>						
			SB	99-811-32-137	0	10/5/1984	99-811-32-137		
<i>Accommodate new LRU P/N 42-757 control unit and update wheel card fail driver test to latest configuration.</i>									
SB	99-811-32-143	0	10/18/1985	99-811-32-143					
<i>Modifies the 99-811 Test Set to more accurately simulate the aircraft ARINC 429 (IRS) transmitter signal</i>									
SB	99-811-32-165	1	7/30/1987	99-811-32-165					
<i>Accommodate new LRU P/N 42- 767 control unit and update wheel card fail driver test to latest configuration.</i>									
SB	99-811-32-209	0	3/1/1991	99-811-32-209					
<i>Accommodate new LRU P/N 42- 767-1 control unit used on the B- 767 aircraft equipped with carbon brakes.</i>									
SB	99-811-32-8	0	12/17/1993	99-811-32-8					
<i>Provides improved wheel speed transducer test.</i>									
SB	99-811-32-9	0	11/1/1996	99-811-32-9					
<i>Provides improved wheel speed transducer test.</i>									
99-815	ANTI-SKID/AUTOBRAKE TEST SET	B737	OM	32-42-94	0	12/21/1984	99-815		
99-873	ANTI-SKID TEST SET	SAAB 340/BAeJ41	O&M	32-42-03	1	5/1/1991	O&M99-873		

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99-887	TBD	B-737-300	O&M	32-42-96	2	9/15/1989	O&M99-887
99-889	ANTI-SKID AUTOBRAKE TEST SET	B-727/737	O&M	32-42-85	2	10/24/1986	O&M99-889
A20080-265	TBD	B-707	OHM	29-20-62	0	9/1/1973	OHM_A20080-265
A60083-CD12	IN-LINE HYDRAULIC RELIEF VALVE	B-707	OHM	29-20-33	0	4/15/1975	OHM_A60083-CD12
A60085-YZ6	IN-LINE HYDRAULIC RELIEF VALVE	B-707/727	OHM	29-20-22	1	8/1/1977	OHM_A60085-YZ6
A60087-CD6	IN-LINE HYDRAULIC RELIEF VALVE	B-707/727/737	OHM	29-20-15	1	3/1/1975	OHM_A60087-CD6
A60205	HYDRAULIC RESERVOIR RELIEF VALVE	B-727	OHM	29-20-09	0	11/15/1975	OHM_A60205
A61095	HYDRAULIC PRIORITY VALVE FREE REVERSE FLOW	DC-8	OHM	29-20-16	0	7/15/1978	OHM_A61095
			TR	TR29-20-16-01	0	1/15/1998	TR29-20-16-01
			<i>(Description not available)</i>				
			TR	TR29-20-16-02	0	1/15/1998	TR29-20-16-02
			<i>(Description not available)</i>				
A61255-1	LINE TYPE THERMAL RELIEF VALVE	B-707	OHM	29-20-34	0	10/1/1974	OHM_A61255-1
A61446-1,-2,-3,-4	NOSE GEAR BY-PASS VALVE	B-727/737	OHM	29-20-03	3	11/15/1992	OHM_A61446-1
A61498	CARTRIDGE TYPE PRESSURE RELIEF VALVE	B-727	OHM	29-20-21	4	5/13/1983	OHM_A61498
A61499	CARTRIDGE TYPE PRESSURE RELIEF VALVE	B-727/737	CMM	29-20-06	5	9/25/1996	CMM61499

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A61671, -1	HYDRAULIC PRESSURE REDUCER VALVE	B-727	OHM	29-20-08	2	7/8/1988	OHM_A61671
A61681	ASYMMETRY SHUTOFF VALVE ASSEMBLY	B-727/737	OHM	29-20-20	0	10/15/1979	OHM_A61681
A61706, 72340	VENTRAL STAIR CONTROL VALVE ASSEMBLY	B-727	OHM	29-20-51	0	12/15/1974	OHM_A61706
A62028,-1	HYDRAULIC PRESSURE REDUCER VALVE	B-727/737	OHM	29-20-07	0	1/1/1975	OHM_A62028
A62193	LOW PRESSURE PRIORITY VALVE	B-707/727/737	OHM	29-20-41	0	12/1/1974	OHM_A62193
A62326	HYDRAULIC PRIORITY VALVE	B-707, 720	OHM	29-20-35	0	10/27/1983	OHM_A62326
			SB	A62326-29-3	0	2/11/1977	A62326-29-3
			<i>Design modification to increase locking force of tab lock nut.</i>				
A62381	FREE REVERSE FLOW HYDRAULIC RELIEF VALVE	B-727	OHM	29-20-05	0	6/1/1975	OHM_A62381
A62395	SHUTOFF VALVE	B-707/727	OHM	29-20-17	1	5/2/1988	OHM_A62395
A62479	LOW PRESSURE PRIORITY VALVE	B-707/727	OHM	29-20-14	1	2/26/1982	OHM_A62479
A63359-1,-2	HYDRAULIC PRIORITY VALVE	F-28/F-228/FOKKER 100	OHM	29-20-56	2	7/15/1991	OHM_A63359-1
A63361, A63363	CARTRIDGE-TYPE RELIEF VALVE	F-28	OHM	29-20-55	1	4/15/1979	OHM_A63361
A65068	HYDRAULIC PRIORITY VALVES	B-707/727	OHM	29-20-04	0	2/15/1975	OHM_A65068
TE-3439	MD-11 AUTOBRAKE CFDS MONITOR	MD-11	CMM	TE-3439	0	1/30/1992	TE-3439

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